

# **Finno Exergy's PGC Technology: a novel solution for efficiency improvement and fuel flexibility in gas turbines**

# FINNO EXERGY

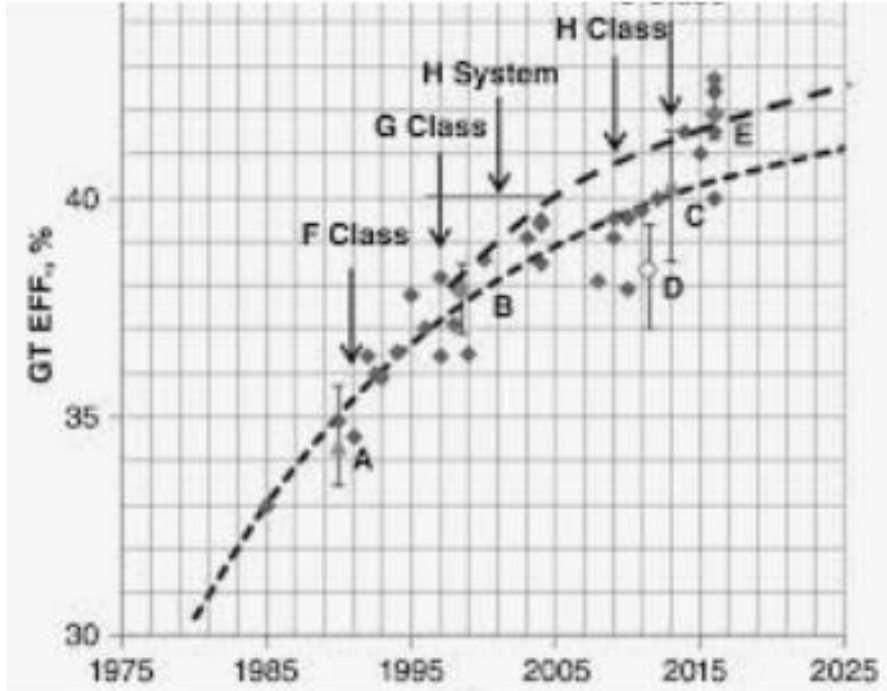
*ENABLING RENEWABLE FUEL  
TRANSITION  
FOR GAS TURBINES*

Winner of the Shell  
New Energy Challenge  
2020



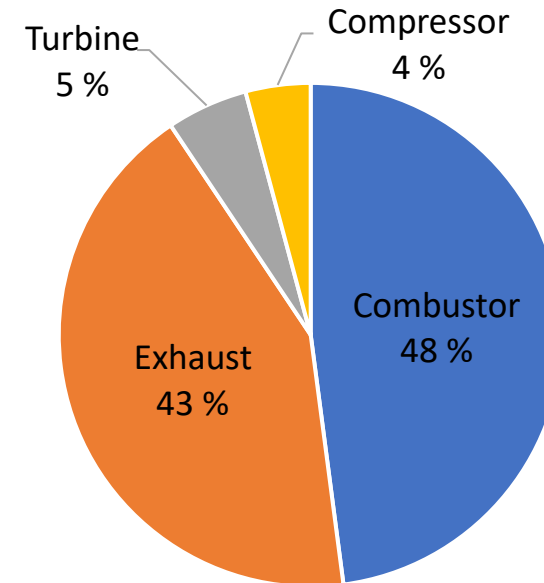
# GAS TURBINE EFFICIENCY AND EXERGY DESTRUCTION

Gas turbine efficiency evolution



Source: Gülen, S. Can. 2019. Gas Turbines for Electric Power Generation. Cambridge: Cambridge University Press.

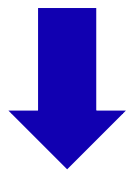
Gas turbine exergy destruction (loss of available energy)



Sources: Elfeituri, I.A., 2017, "Exergy Based Performance Analysis of a Gas Turbine Unit at Various Ambient Conditions", International Journal of Energy and Power Engineering

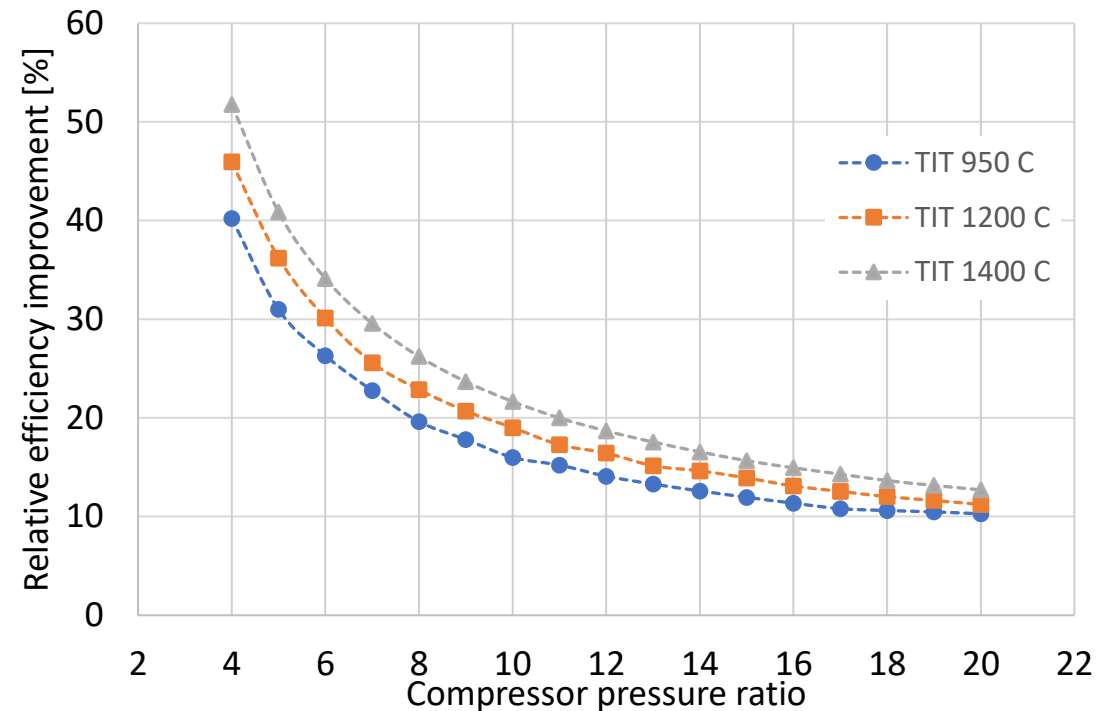
# PRESSURE GAIN COMBUSTION

- Replacing Isobaric Heat Addition with Isochoric Heat Addition (**Pressure Gain Combustion**) will greatly reduce entropy
- From continuous combustion to **periodic combustion**.
- For the same compressor pressure ratio and heat input, **higher turbine inlet pressure**.
- **More power** can be extracted from turbine, for the same amount of fuel burned.



**Higher thermal efficiency!**

## Potential gas turbine efficiency improvement due to PGC\*

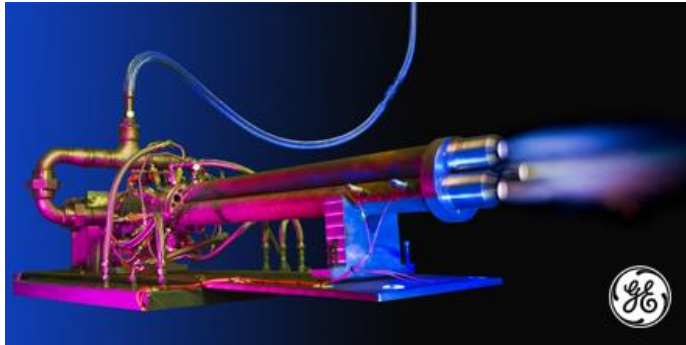


**FINNO EXERGY**

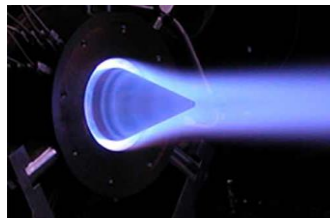
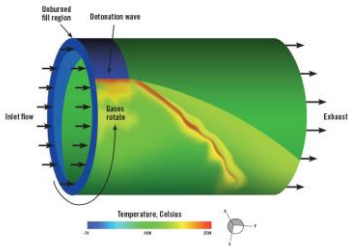
\*Finno Exergy's calculation

# PRESSURE GAIN COMBUSTION

Pulsating detonation engine

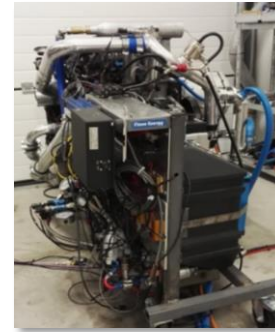


Rotating Detonation Engine

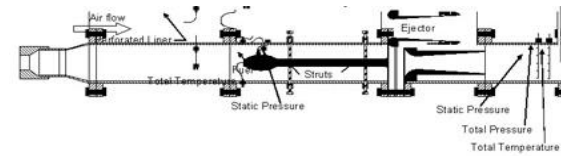


DETONATION

DEFLAGRATION



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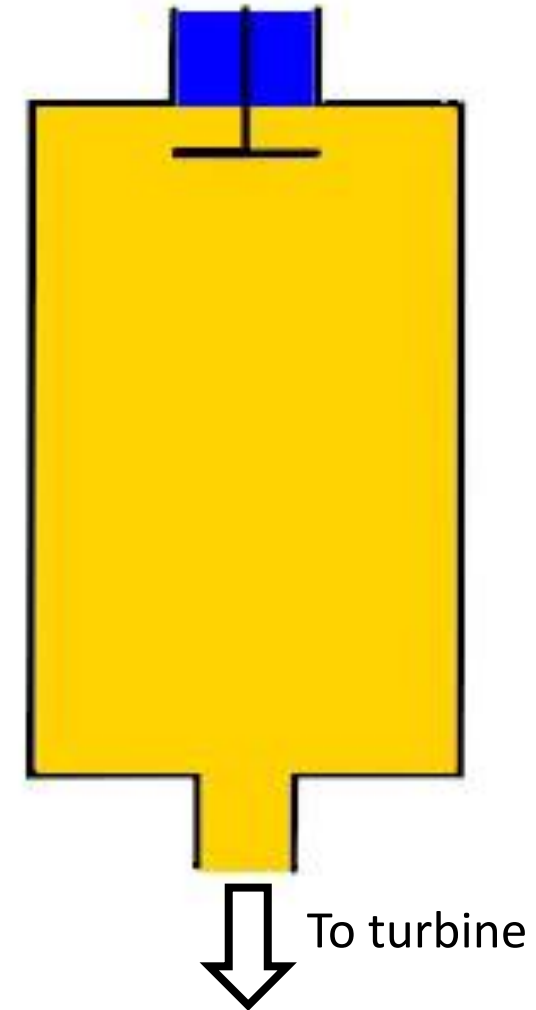
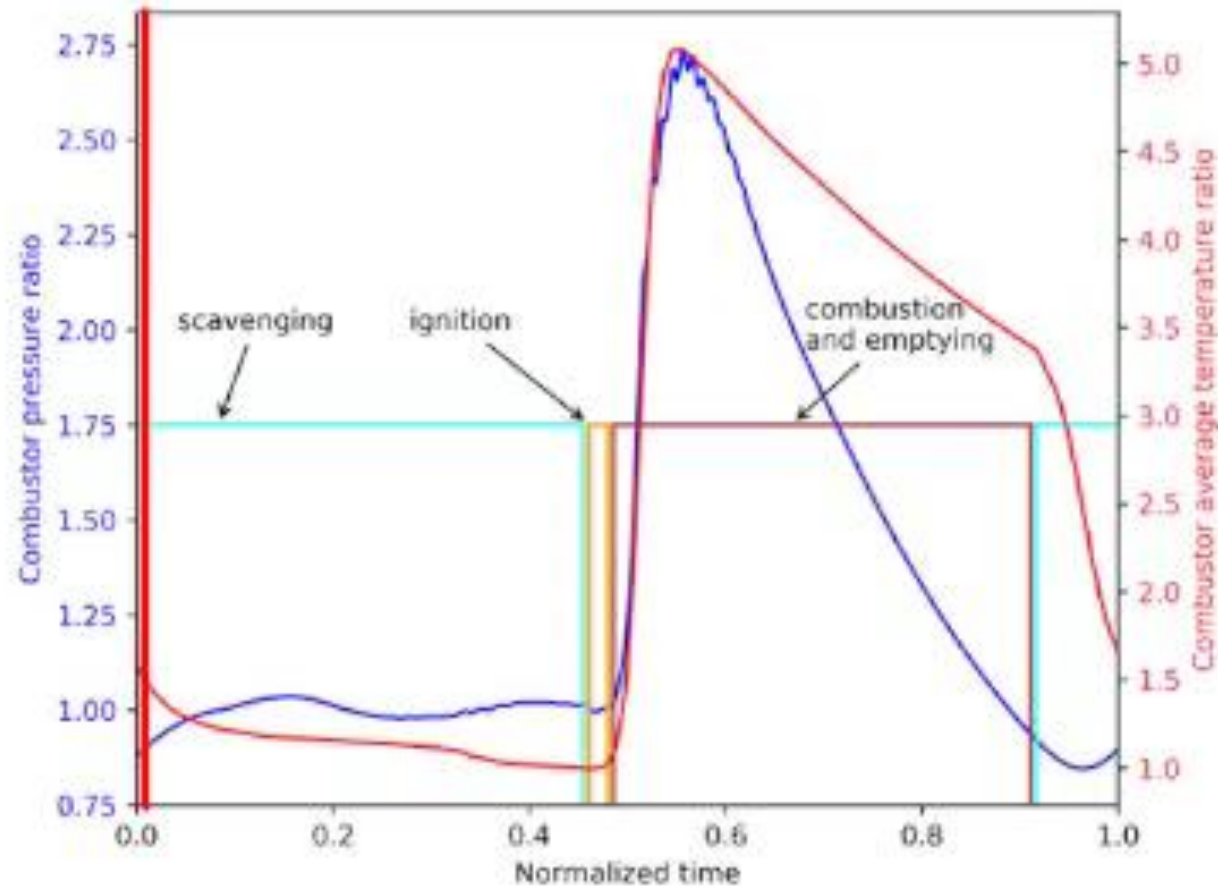
Pulsejet



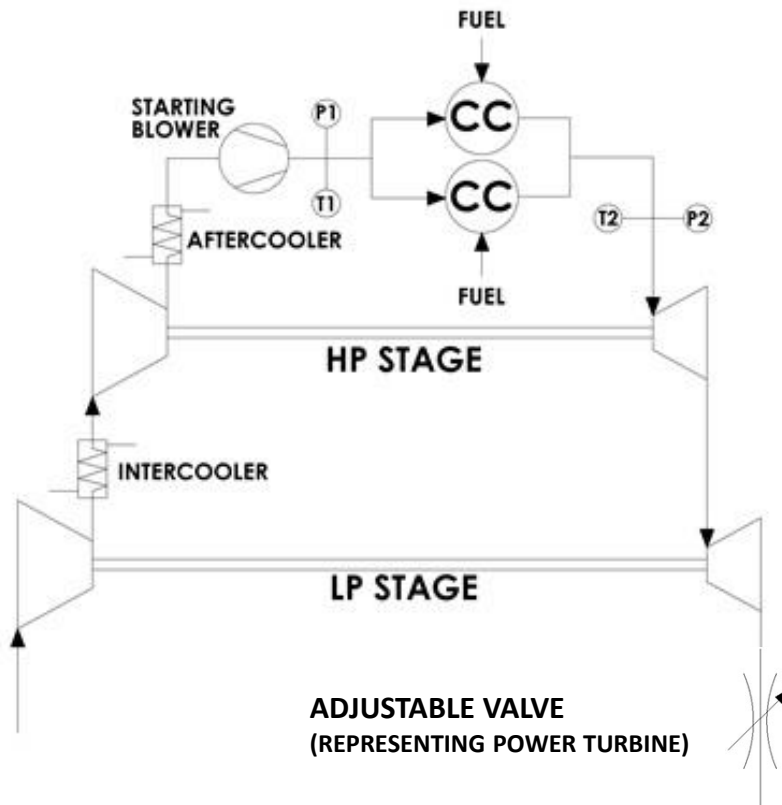
Constant volume combustion

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# FINNO EXERGY'S PGC TECHNOLOGY BASIC WORKIN PRINCIPLES



# PROVE OF CONCEPT TEST UNIT: EXPERIMENTAL FACILITY DESCRIPTION

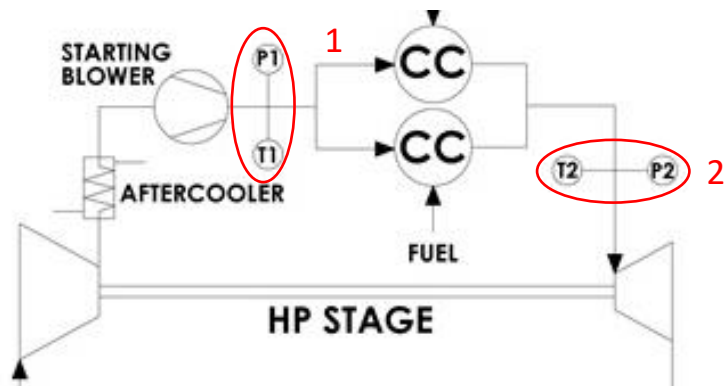


Main operative parameters for a typical test

Frequency [Hz]	15
Test duration [min]	3-5
Combustor pressure inlet [bar]	4
Turbine inlet temperature [K]	1193
Combustor Temperature Ratio	≈3.2
Air-Fuel Equivalent Ratio ( $\lambda$ )	2.0-2.8
Fuel Power [kW]	≈150
Automotive turbochargers type for LP and HP stages: (Mitsubishi)	TD02-TD04

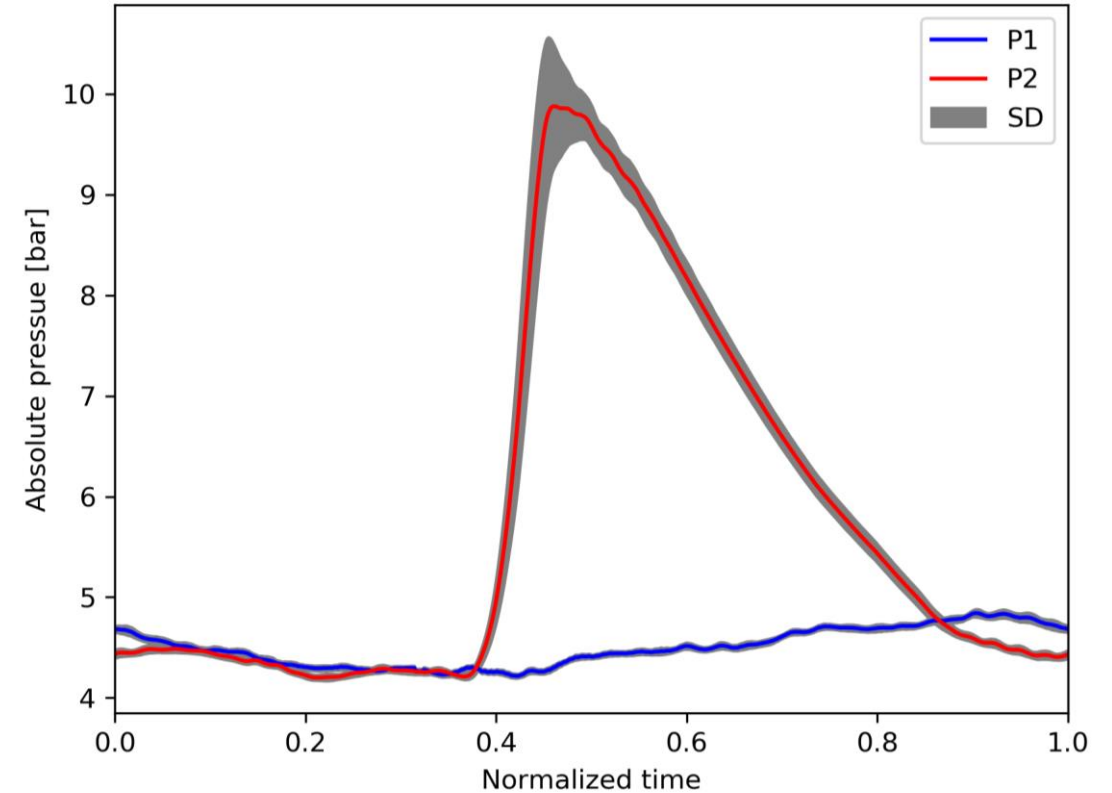
# TEST UNIT: EXPERIMENTAL RESULTS ON PRESSURE GAIN CAPABILITY

Operational frequency [Hz]	15
Logging frequency [Hz]	10800
N° of consecutive cycles	300
T <sub>1</sub> [K]	366
T <sub>2</sub> [K]	1222
P <sub>1</sub> [bar]	4.50
P <sub>2</sub> [bar]	5.75
Pressure Gain* [%]	27.9



Highlight of measurement stations 1 and 2

[Experimental Demonstration of a Novel Deflagration-based Pressure Gain Combustion Technology | AIAA Propulsion and Energy Forum](#)



**Cycle average P1 and P2**

The red and blue curves are the result of averaging the pressure measured in each point in 300 consecutive cycles. Standard deviation in each point is represented as the grey area in the plot

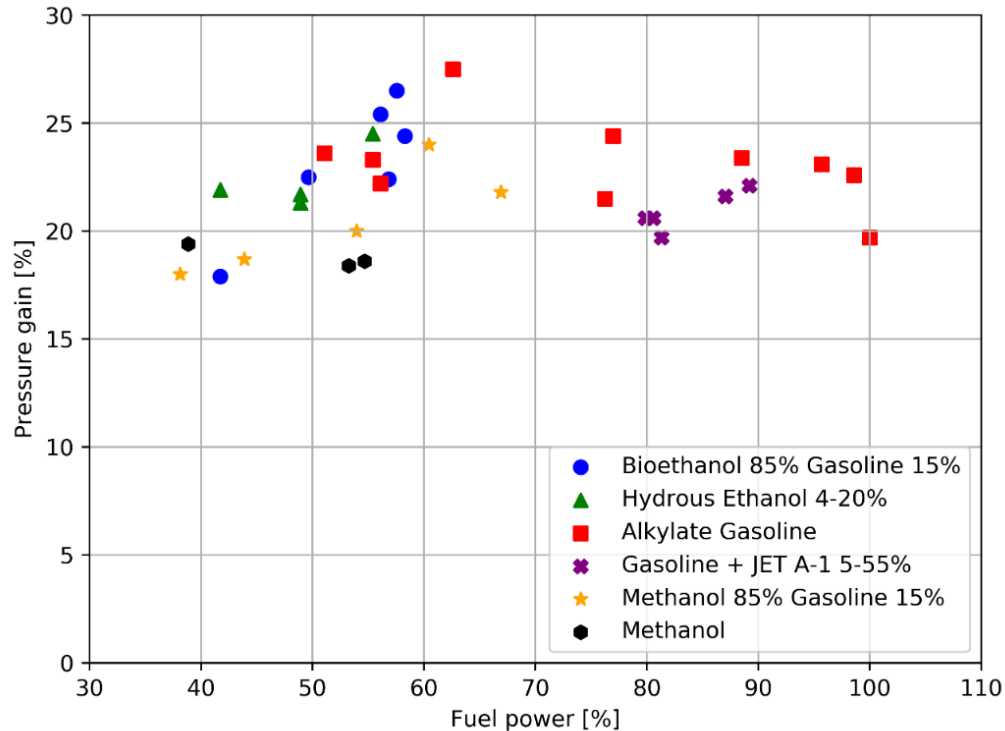
$$* PG\% = \left( \frac{P_2}{P_1} - 1 \right) * 100$$

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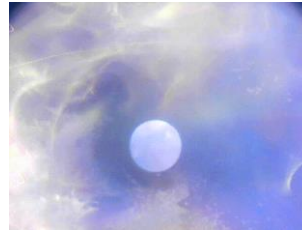
# FUEL FLEXIBILITY AND HYDROGEN

## Data collected from the test unit

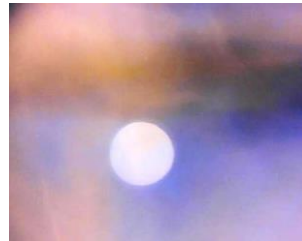


Able to sustain pressure gain level over 20% with different setups, operating points, and types of fuel

## Inside the combustor



Bioethanol



Methanol



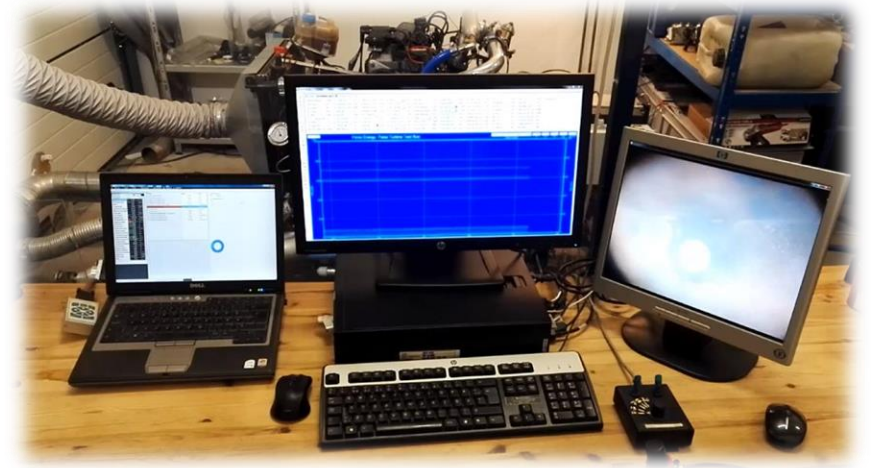
Alkylate gasoline

## Perfectly suited for hydrogen

- Pressure gain combustion technology are commonly operated on hydrogen.
- Finno Exergy's combustion system contains functions which enables extreme fuel flexibility



# NEW ENERGY CHALLENGE

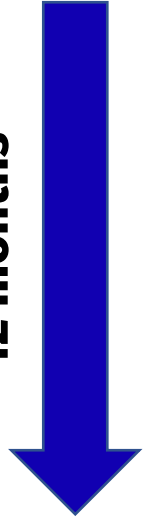


**Finno Exergy won the Shell New Energy Challenge  
2020**

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# NEW ENERGY CHALLENGE 2021: PROJECT RESULTS

12 months



- Extract power from the test unit ✓
- Measure turbine efficiency and identify eventual losses ✓
- Optimize the system for highest efficiency and compare it with an equivalent traditional gas turbine ✓

**PROVE HIGHER GAS  
TURBINE EFFICIENCY  
IN A SMALL-SCALE  
APPLICATION**



**39.5% MORE EFFICIENT**

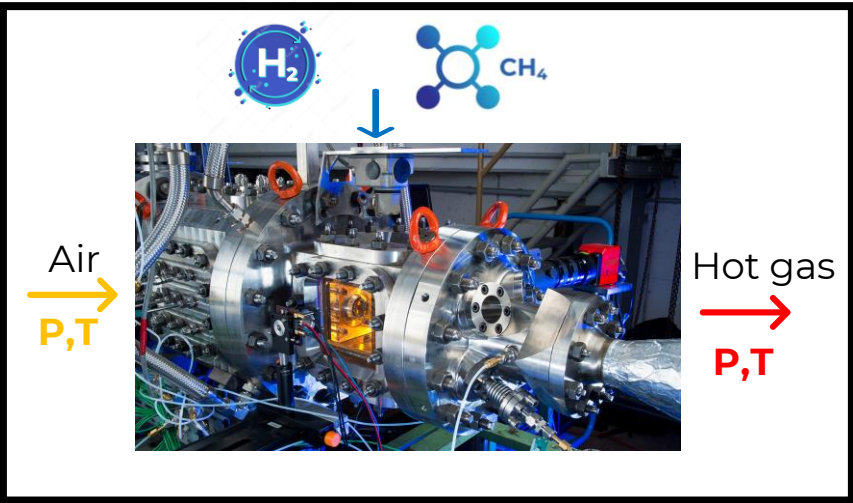
# SHELL GAME CHANGER PROJECT

18 months

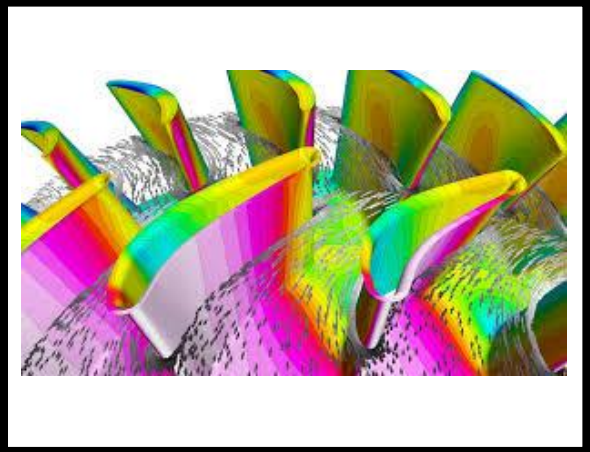
- Project planning and design of new test rig
- Combustor test in new lab in Finland
- Retrofitting analysis and start commercial discussions with GT OEMs

**PROVE HYDROGEN  
METHANE FUEL FLEX**

**TECHNOLOGY  
DEVELOPMENT ROADMAP  
FOR RETROFITTING**



**TEST RIG IN HELSINKI**



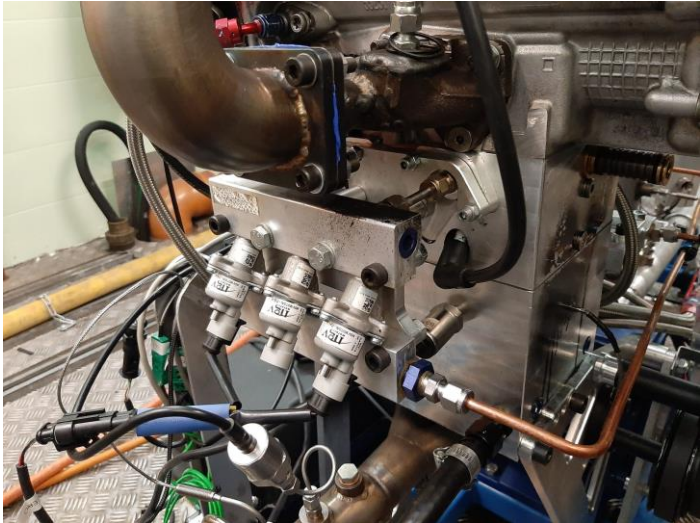
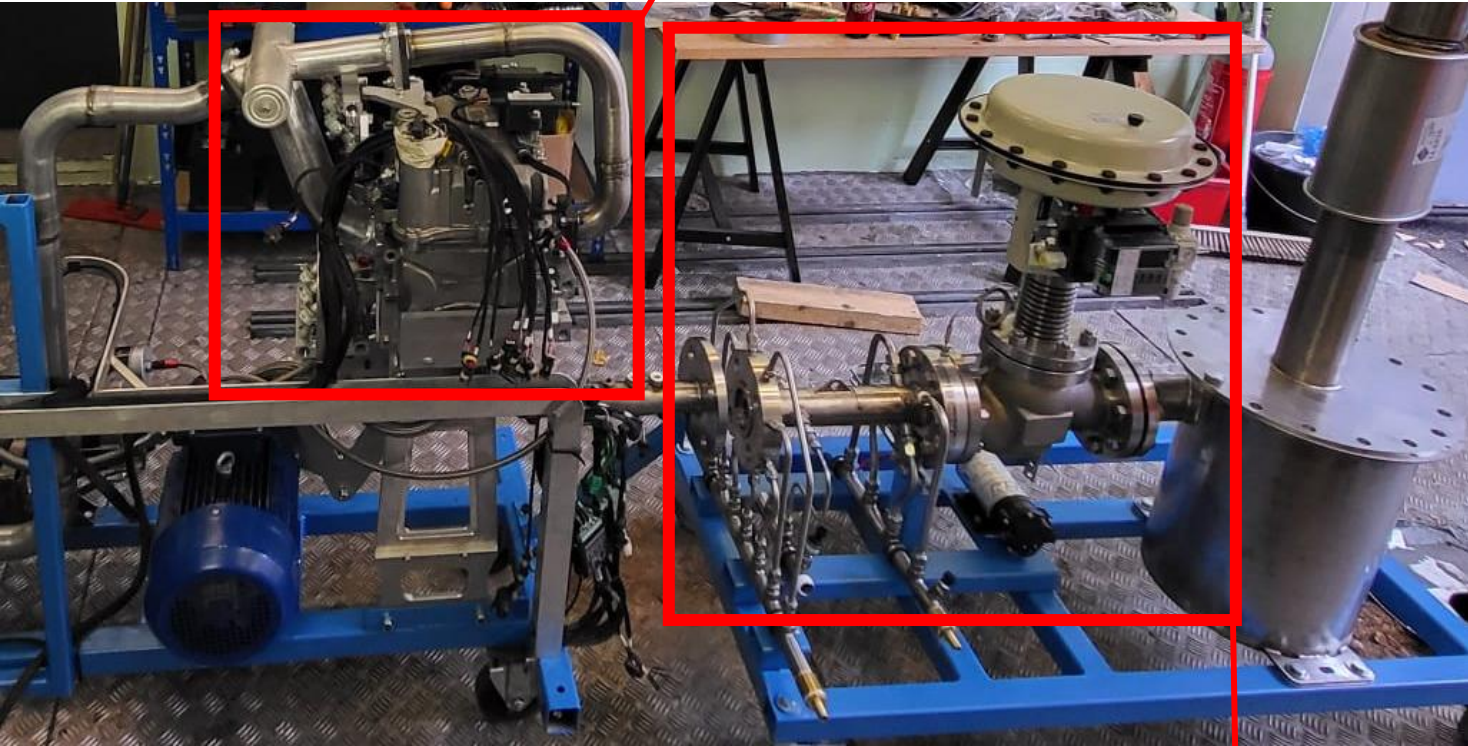
**RETROFITTING  
SIMULATIONS**

**START COMMERCIAL  
RELATION WITH GT  
OEMs**

# COMBUSTOR AND TEST RIG UPGRADE

New test rig

Improved combustor design



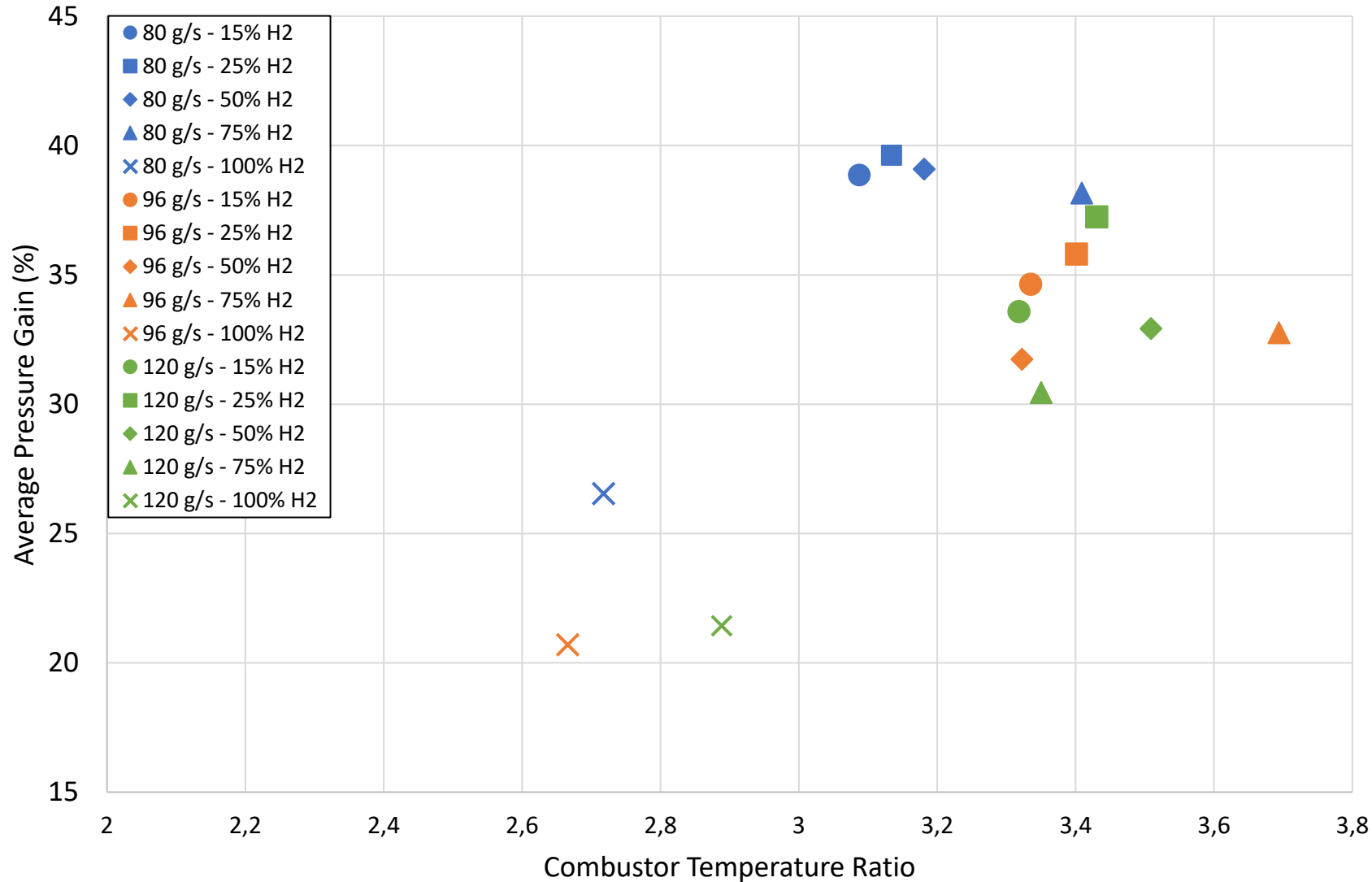
New H2 and CH4 fuel system

New hot section to simulate the presence of a GT



# EXPERIMENTAL TEST RESULTS

Pressure Gain against Temperature Ratio for varying air mass flow and fuel blends



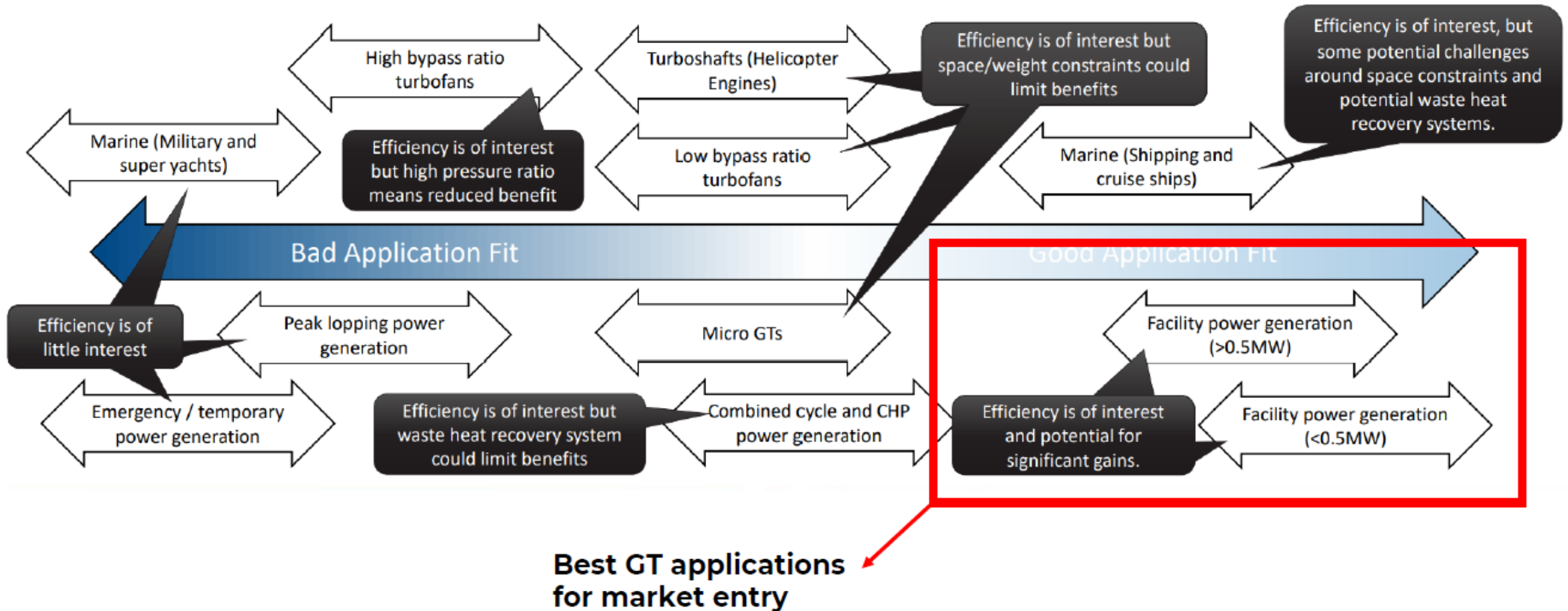
An ideal industrial gas turbine with or without FE PGC system

	FE- PGC	Brayton
Compressor work [kW]	50,7	50,7
Heat input [kW]	100,8	100,8
Extracted turbine's work [kW]	87,0	81,1
Turbine efficiency	0,88	0,88
System thermal efficiency	36,0	30,2

**Potential for 20% reduction in fuel consumption**

**FINNO EXERGY**

# MARKET STUDY RESULTS





# MAJOR TECHNICAL RISK FOR RETROFITTING

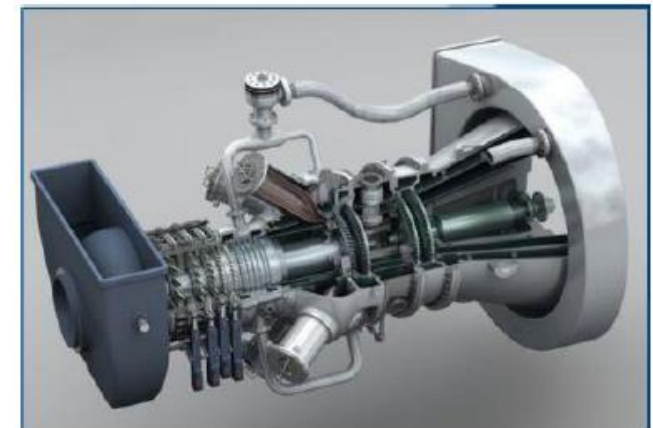
## Integration challenges

- Mechanical vibration (**High Risk**)
- Axial load balancing (**High risk**)
- Bearings life and lubrication (**Medium Risk**)
- Turbine cooling (**Medium Risk**)
- Compressor surge and stall (**Medium risk**)
- Reliability of valves and ignition sources and fuel injectors (**Medium Risk**)
- Material degradation (**Medium Risk**)
- Turbine performance (**Low Risk**)

Best GT candidates for first retrofit (examples)



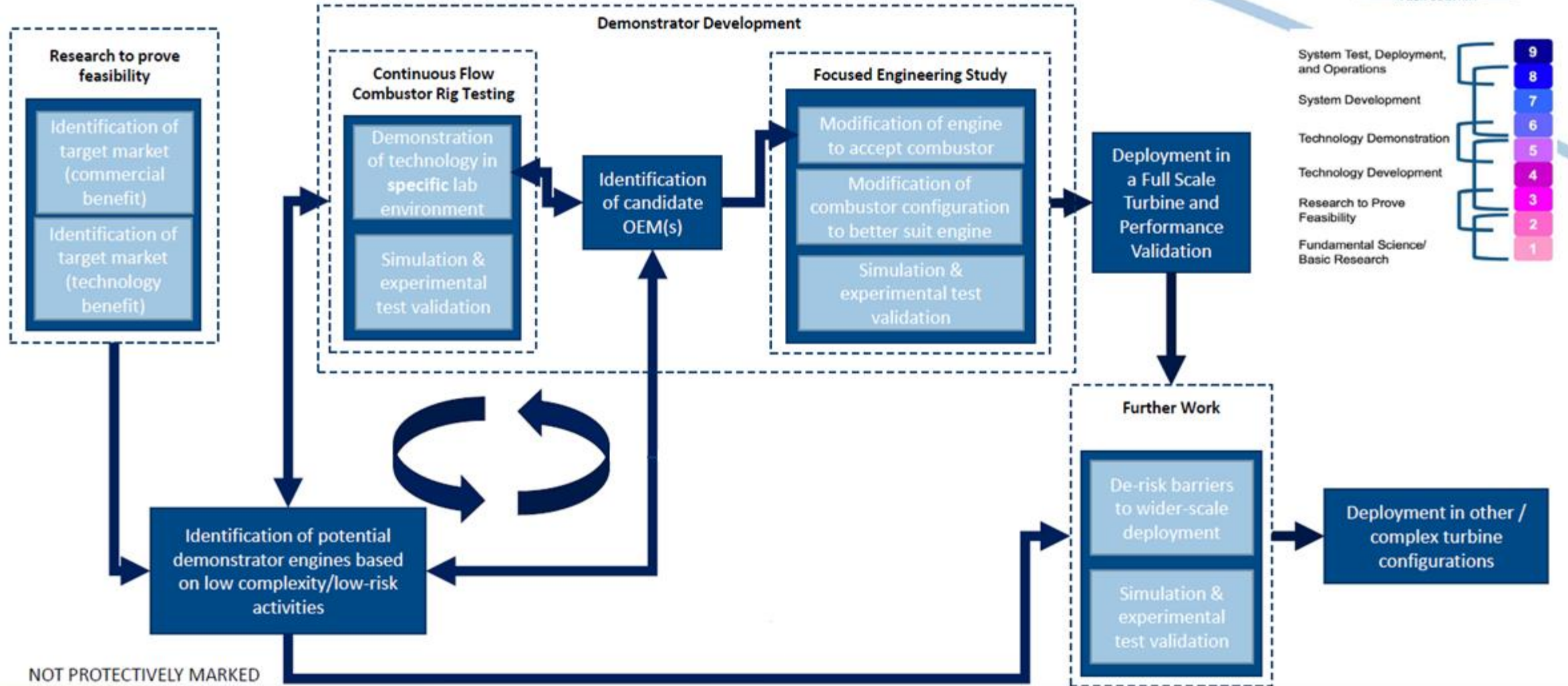
**Solar Saturn**



**MAM MGT6000**

# TECHNOLOGY DEVELOPMENT ROADMAP

## Overall Road Map Development Plan



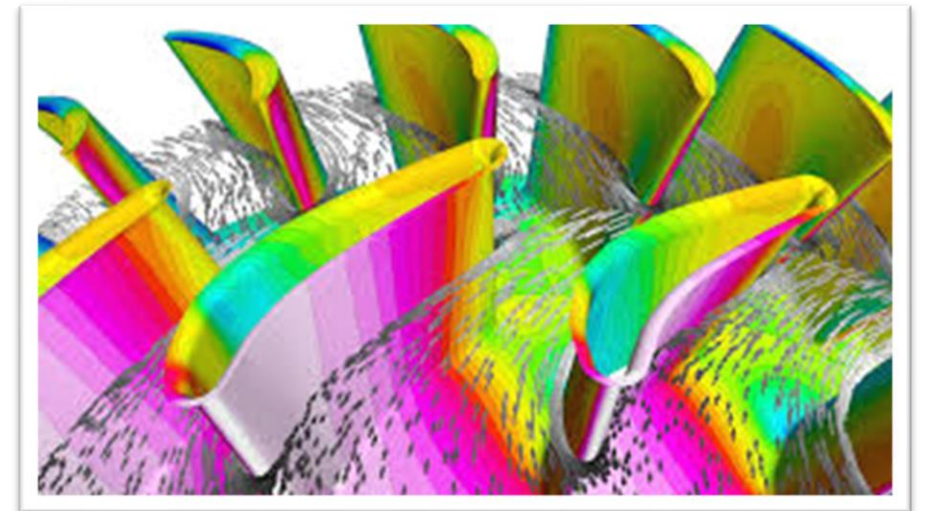
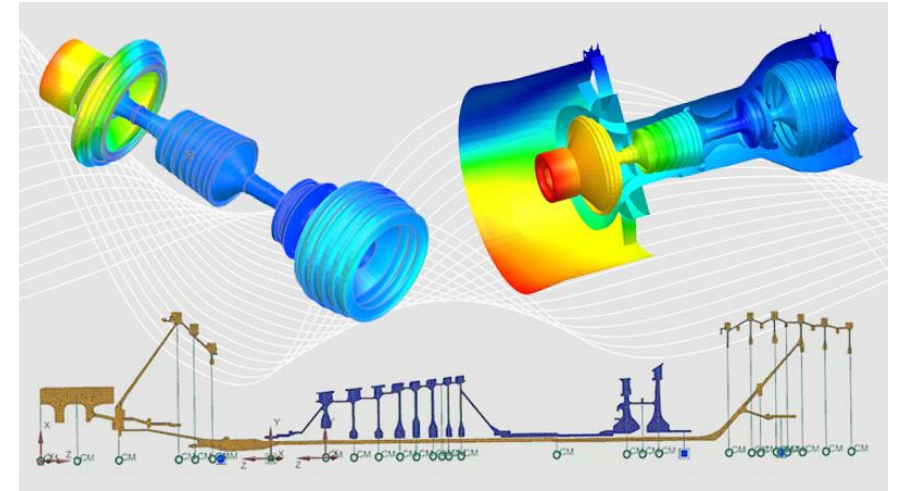
# NEXT TECHNOLOGY DEVELOPMENT PROJECT

**FINNO IS STARTING A NEW PROJECT IN COLLABORATION WITH SHELL, GAS TURBINE'S OEMS, UNIVERSITIES AND SPECIALIZED RESEARCH CENTER**

**The project will use rotordynamic models, 1D and 3D CFD models to simulate the effect of integrating Finno Exergy's PGC system in one selected industrial Gas Turbine.**

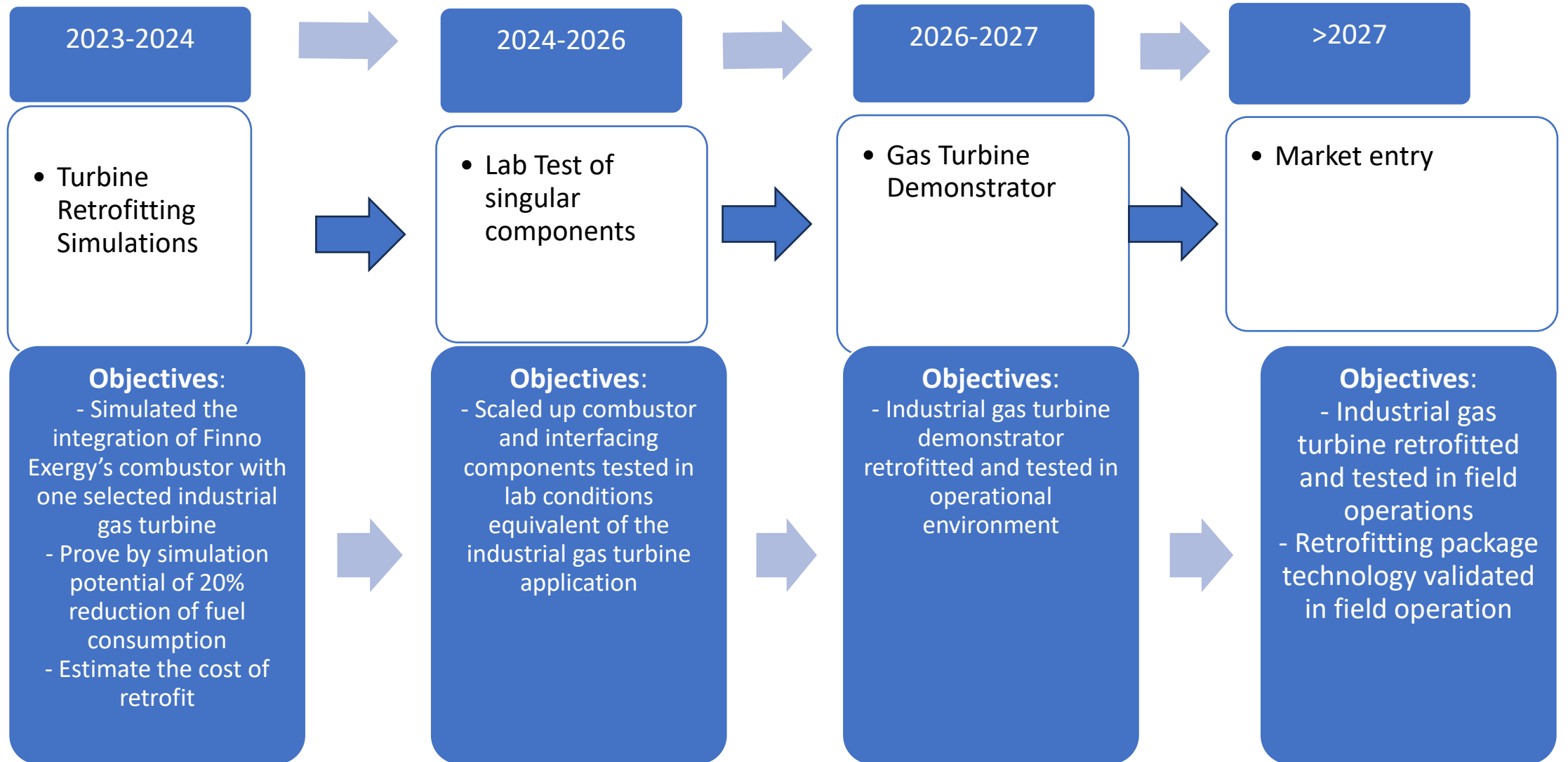
**The objectives will be:**

- **Calculate overall GT efficiency improvement / power increase**
- **Verify mechanical integrity of the retrofitted GT**
- **Calculate cost of GT retrofit**



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# LONG TERM DEVELOPMENT PLAN



**Thank you for your attention**