

RISK-BASED APPROACH TO ASSESS TECHNO-ECONOMIC FEASIBILITY OF GAS TURBINE COMPONENTS LIFE EXTENSION

Jean-Louis Meyer¹, Salah Morsili¹, Alfonso Pandolfi², Ugo Simeoni³,
Marco Sorrentino²

msorrentino@unisa.it

¹Électricité de France (EDF)

²Department of Industrial Engineering (DIIN)- University of Salerno

³ETN Global

Introduction

The aim of this work is to propose a purely **data-driven approach** to maintenance.

This need arises for **two main reasons**:

- The preventive maintenance intervals set by OEMs sometimes prove to be **inaccurate**, leading to considerable increases in **maintenance costs**.
- The physical modelling proposed in many works, even if it can provide accurate answers, is very **complex** and **specific** and often requires **costly analysis** and plant **downtime**.

Gas Turbine degradation

When a gas turbine system is put into operation, an expected "natural" degradation is attributed to it. On the basis of the **operating history**, the initially expected value is updated.

It is clear that if a failure occurs, the **useful life** of the system undergoes a **faster reduction** than the "natural" expected.

It is therefore necessary to identify a failure as soon as possible in order to take **mitigation actions** as efficiently as possible

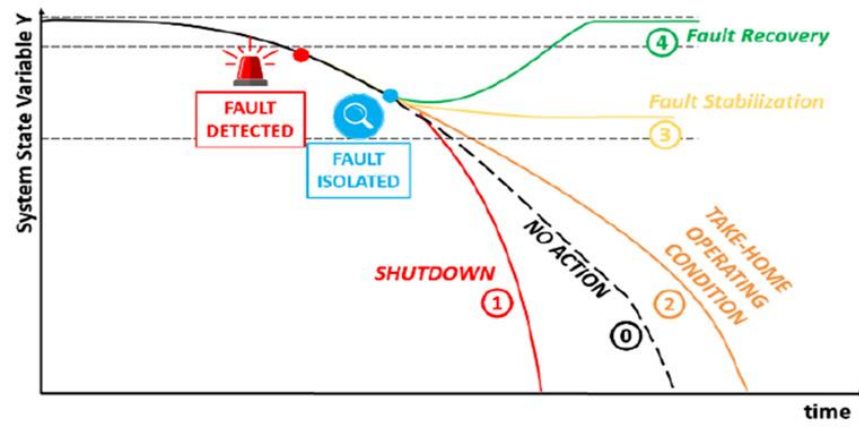


Figure 1: Fault mitigation approach [1]

Literature analysis

Literature analysis

- Study of the main **damage mechanisms**, maintenance policies and the **state of the art**.

Definition of the methodology

- Data-driven **methodology** that provides alternative solutions.

Testing & validation

- **Application** of methodology on a dataset.

Key Damages

The main damage mechanisms involved in gas turbine systems are:



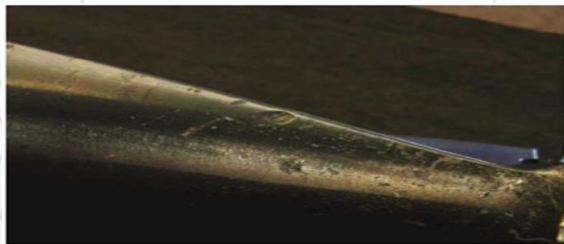
FOD



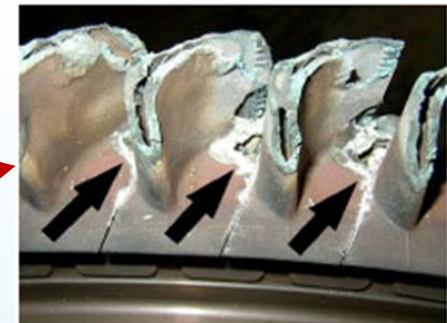
CREEP

FOULING

EROSION



CORROSION



FATIGUE

State of art

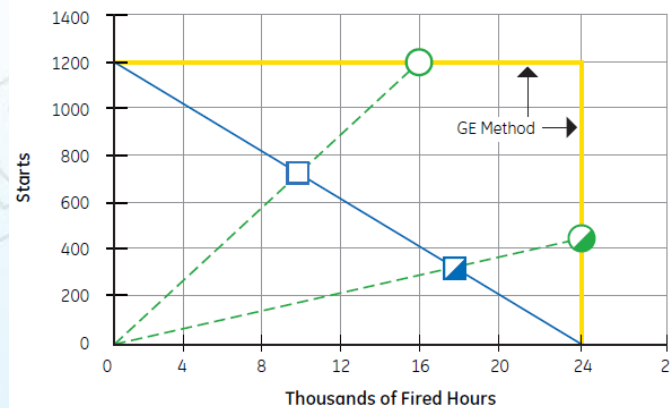
The most widely used approach by the OEMs to define the life reduction of a component is to establish **equivalent operating hours (EOH)**.

The most general **formula** to calculate the EOH is as follow [2]:

$$EOH = a_1 * n_1 + a_2 * n_2 + \sum_{i=1}^n t_i + f * w * (b_1 * t_1 + b_2 * t_2)$$

- a_1 and n_1 are the coefficient and number of starts, respectively.
- a_2 and n_2 are the coefficient and number of emergency starts, respectively.
- b_1 and t_1 are coefficient and operating time in basic load, respectively.
- b_2 and t_2 are coefficient and operating time of the peak load, respectively.

GE bases gas turbine maintenance requirements on independent counts of starts and hours.



Impact of operating history

The following table makes a distinction between the most common faults in systems that perform **continuous operations** and systems that perform **cyclical operations**.

Continuous Duty	Cyclic Duty
Creep Oxidation Corrosion Erosion FOD Rupture Rubbing/Wear High-Cycle Fatigue Combined failure mechanism (creep/fatigue, corrosion/fatigue, oxidation/erosion and so on)	Thermal-Mechanical Fatigue High-Cycle Fatigue Rubbing/Wear FOD Combined failure mechanism (creep/fatigue, corrosion/fatigue, oxidation/erosion and so on)

Table II: Typical failure modes for hot gas components inside a gas turbine [3]

Definition of the methodology

Literature analysis

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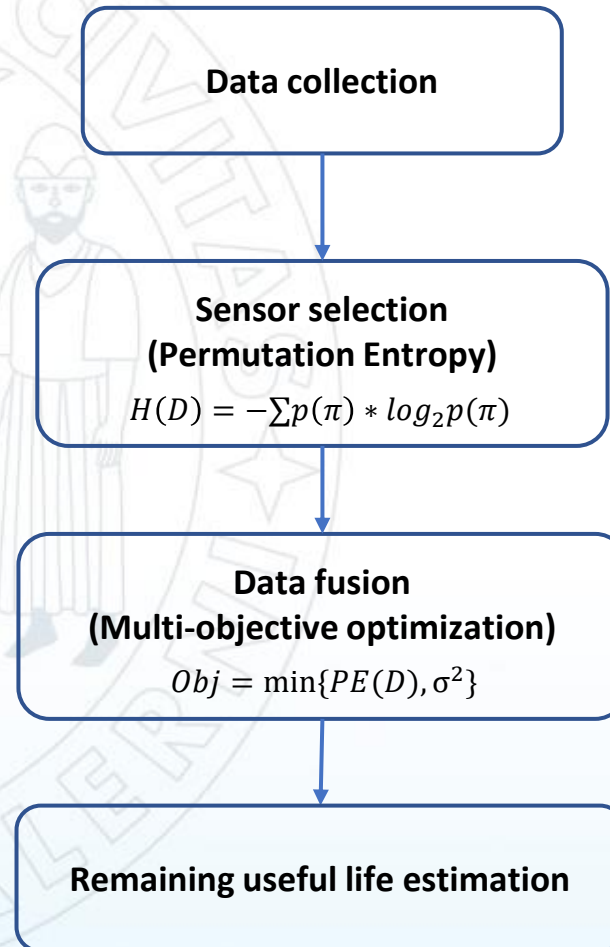
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Methodology



Permutation Entropy

PE is a measure of complexity of a dynamic system based on comparison of neighbouring values [9].

Probability of each permutation:

$$p(\pi) = \frac{\#\{t | 0 \leq t \leq T - n, (x_{t+1}, \dots, x_{t+n}) \text{ has type } \pi\}}{T - n + 1}$$

Permutation Entropy calculation:

$$H(n) = -\sum p(\pi) * \log_2 p(\pi)$$

Divide all by $\log_2 n!$ to normalise

$$0 \leq H(n) \leq \log_2 n!$$

How it works 1/3

To give an idea of how PE is calculated [10], an **example** of calculation on the following vector x is provided:

$$x = [4 \ 7 \ 9 \ 10 \ 6 \ 11 \ 3]$$

Embedding time delay $\tau = 1$

Embedding dimension $n = 3$

$$\begin{bmatrix} 4 & 7 & 9 & 10 & 9 \\ 7 & 9 & 10 & 6 & 11 \\ 9 & 10 & 6 & 11 & 3 \end{bmatrix}$$

All columns have 3 elements (since $n = 3$) and all the column vectors are one step ahead of the previous one (since $\tau = 1$)

How it works 2/3

The next step is to calculate $3! = 6$ **permutations** and collect them into vectors.

$$\pi_1 = [0 \ 1 \ 2]$$

$$\pi_2 = [0 \ 2 \ 1]$$

$$\pi_3 = [1 \ 0 \ 2]$$

$$\pi_4 = [1 \ 2 \ 0]$$

$$\pi_5 = [2 \ 0 \ 1]$$

$$\pi_6 = [2 \ 1 \ 0]$$



$$\begin{bmatrix} 4 & 7 & 9 & 10 & 9 \\ 7 & 9 & 10 & 6 & 11 \\ 9 & 10 & 6 & 11 & 3 \end{bmatrix}$$




$$\begin{bmatrix} 0 & 0 & 1 & 1 & 1 \\ 1 & 1 & 2 & 0 & 2 \\ 2 & 2 & 0 & 2 & 0 \end{bmatrix}$$

$\pi_1 = [0 \ 1 \ 2]$ because $4 < 7 < 9$

How it works 3/3

<i>Permutation</i>	<i>Occurrences</i>	<i>Relative Frequency (p_i)</i>
π_1	2	2/5
π_2	0	0/5
π_3	1	1/5
π_4	2	2/5
π_5	0	0/5
π_6	0	0/5


$$PE = - \sum_{i=1}^{n!} p_i * \log_2 p_i \approx 1,522$$

With $n=2$ $PE = 0,918$

Data Fusion

The sensors selected based on the sensor selection algorithms can be used in **data fusion**, which offers the following advantages:

- It allows to obtain a health indicator with more **evident trends**.
- It allows us to observe **multiple sensors** simultaneously.
- By calculating the weights on a limited number of the latest observations, it is possible to **update it** continuously.

Fused indicator calculation

The objective of data fusion is to calculate a **vector of weights** which when multiplied by the **matrix containing the chosen sensors** allows us to obtain a fused indicator.


$$\begin{bmatrix} \text{Selected sensor \#1} \\ \text{Selected sensor \#2} \\ \vdots \\ \text{Selected sensor \#n} \end{bmatrix} \begin{matrix} * \\ \\ \\ \\ \end{matrix} \begin{bmatrix} \text{Weights vector} \end{bmatrix} = \begin{bmatrix} \text{Fused indicator} \end{bmatrix}$$

$m \times n$ $n \times 1$ $m \times 1$

The vector of weights is calculated from the matrix of selected sensors with the aim of **minimizing objective functions**.

Multi-objective optimization problem

The objective functions to be minimized in this specific case are two: **variance** and **permutation entropy** [11].

$$Obj = \min_w \{PE(D), \sigma^2\}$$

The variance is calculated according to the following formula:

$$\sigma^2 = \frac{\left(Y_w - \frac{1_M Y_w}{M}\right)' \left(Y_w - \frac{1_M Y_w}{M}\right)}{M - 1}$$

Where Y is the matrix of selected sensors, but it only contains a **limited number** of the latest observations. M is the number of units.

This allows the algorithm to update itself and follow new trends in the data [1].

RUL estimation 1/2

The following formula describes what the RUL is and how it is calculated:

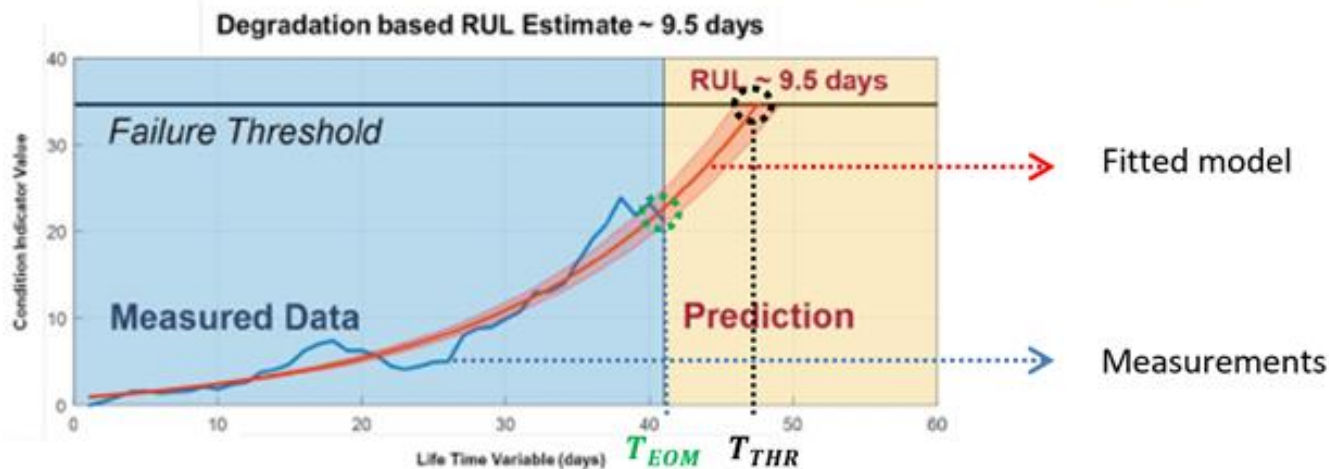
$$RUL = T_{THR} - T_{EOM}$$

where T_{EOM} (**end of measurements time**) is the time corresponding to the end of the available measurements and T_{THR} (**threshold time**) is when the predicted degradation curve meets the threshold.

The method used to calculate RUL **depends on the kind of data available.**

RUL estimation 2/2

In most cases the only data available are on **prescribed threshold values**. With this kind of information, it is possible to fit time series models to **condition indicators** extracted from **sensor data**, which rise or fall over time. These degradation models estimate RUL by predicting **when the condition indicator will cross the threshold**.



Testing & validation

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Testing & validation

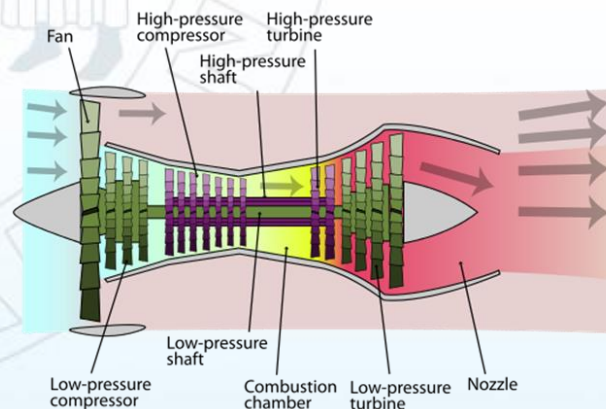
- **Application** of methodology on a dataset.

Dataset description

Measurements of 21 sensors from an **aircraft engine** are taken from the NASA database [12].

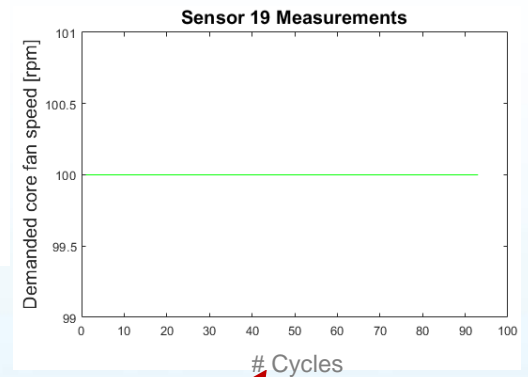
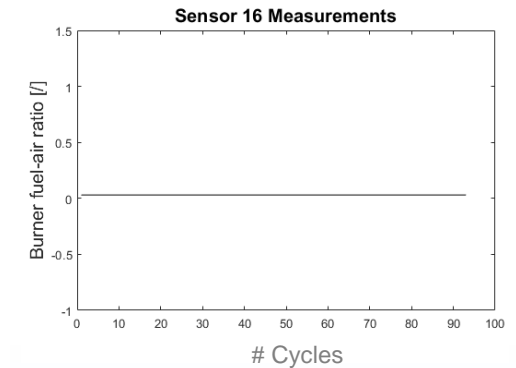
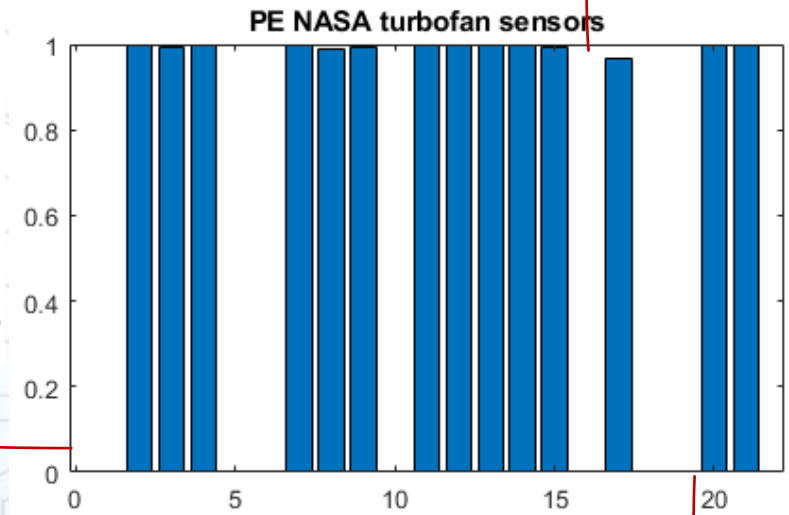
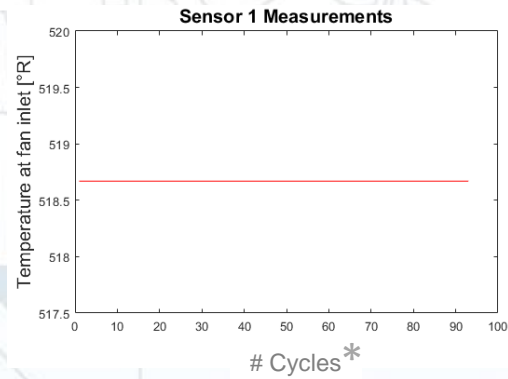
These data were generated with a simulation tool and are available online. Each data set is comprised of a time series of flight cycle measurement “**snapshots**” at **cruise conditions** [13].

Data from the **first fleet** (divided into 21 sensors) were used for this analysis.



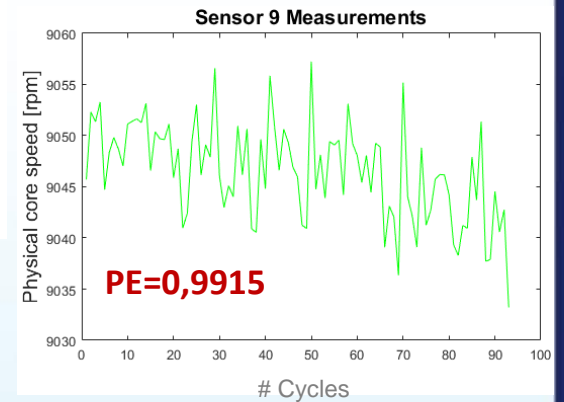
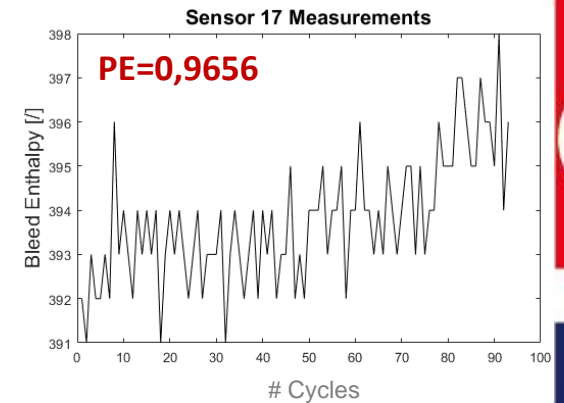
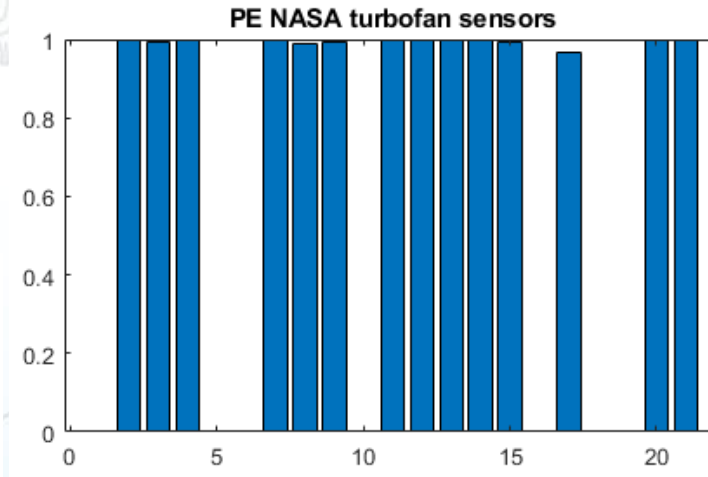
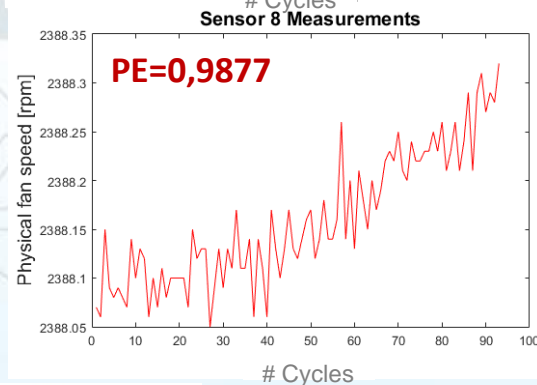
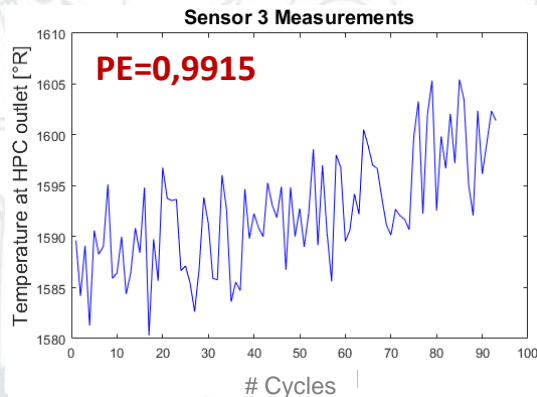
PE calculation - Discarded sensors

Sensors #1, #5, #6, #10, #16, #18 and #19 show a zero PE value justified by the fact that these sensors have a **constant trend**.

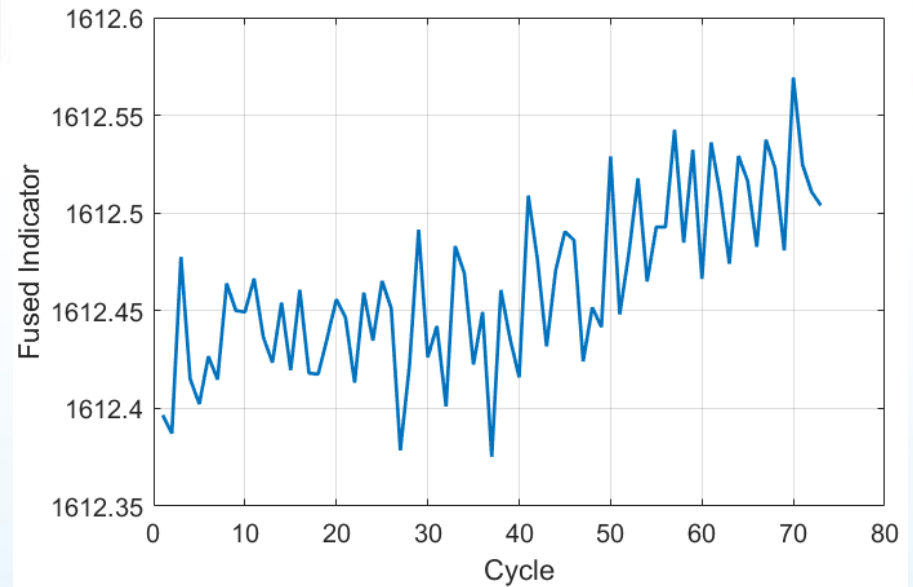
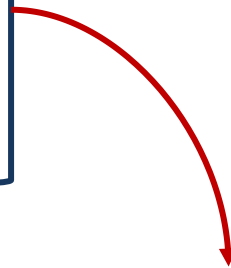
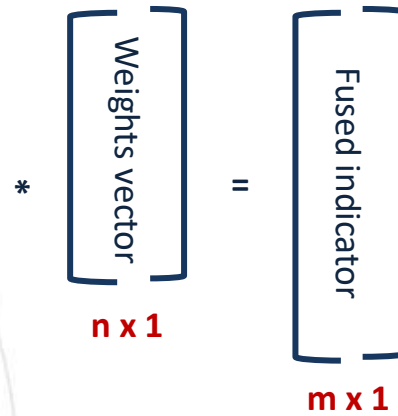


PE – Selected sensors

Sensors #3, #8, #9 and #17 show the lowest PE values. Therefore, they potentially represent the measurements of **highest interest** in a degradation analysis.



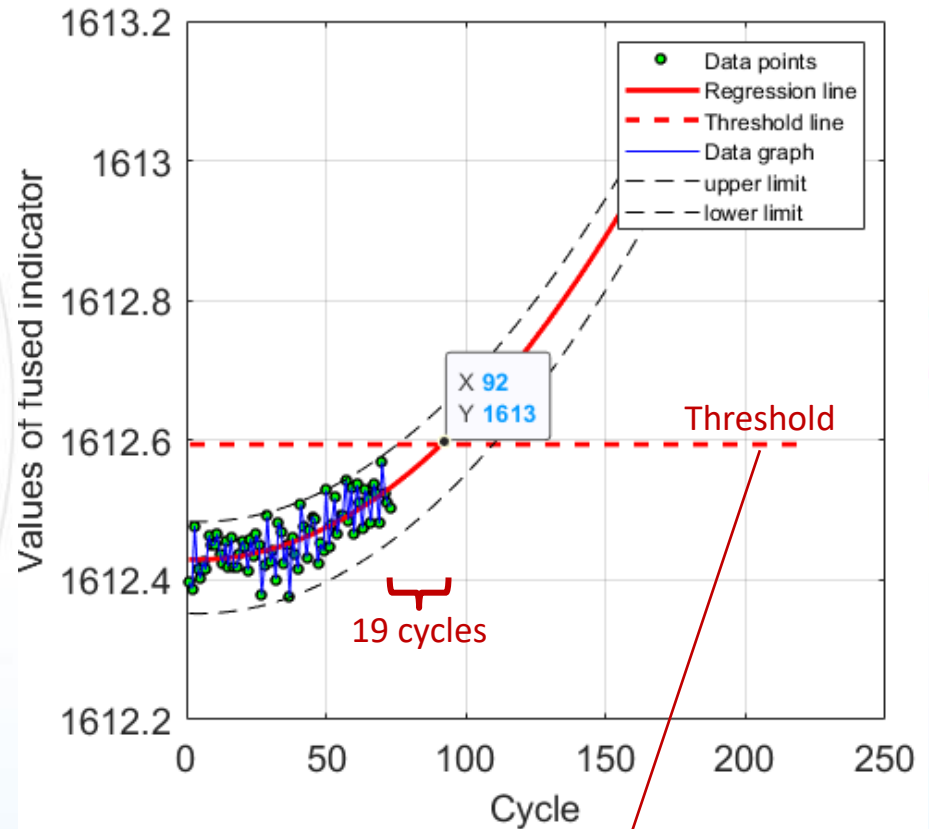
Fused indicator calculation



Selected sensors	#3	#8	#9	#17
weights	0,0022	0,6598	0,0032	0,0115

Remaining useful life

The prediction made on the fused indicator shows a remaining useful life (RUL) of **19 cycles**. Failure occurs at the 92nd cycle and the **effectiveness** of this prediction is therefore confirmed.

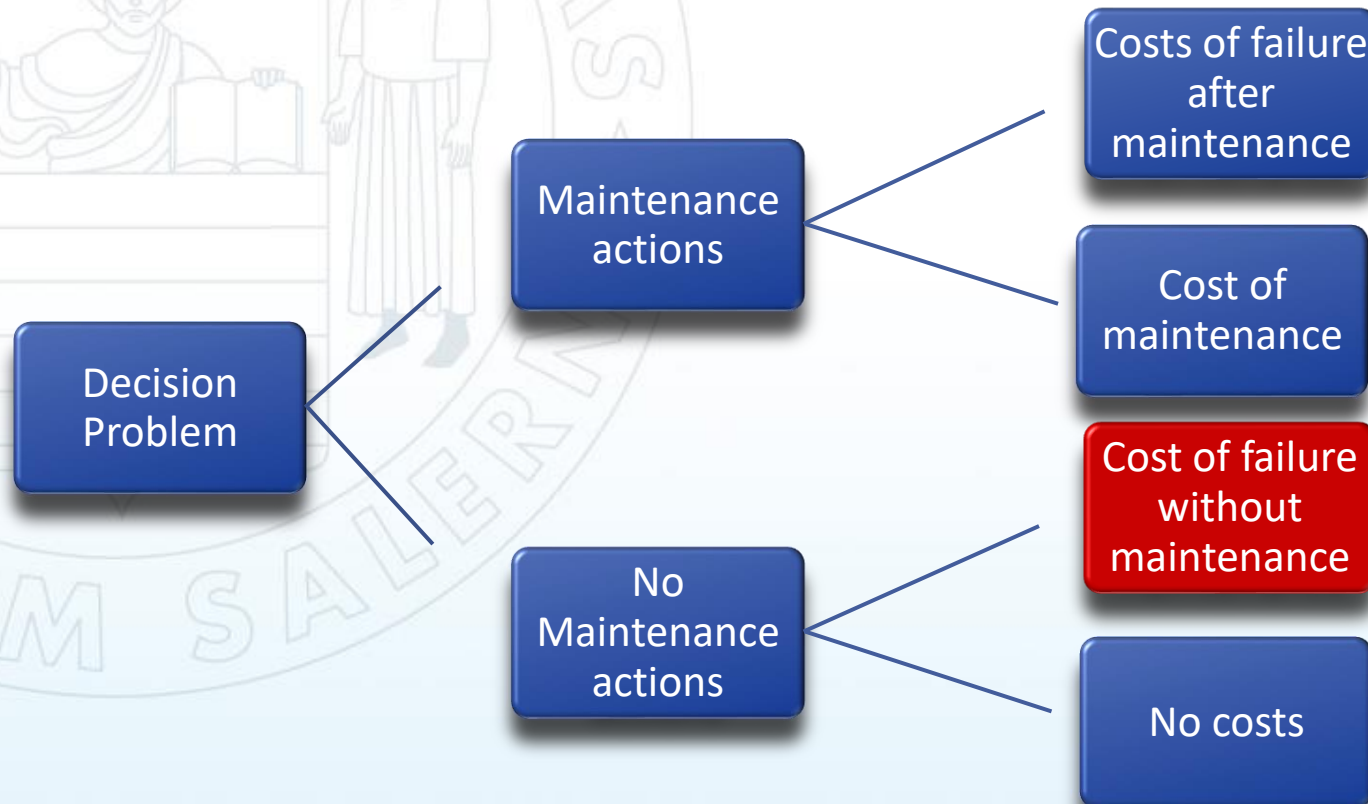


*Threshold Values * Weights = Fused Indicator Threshold*

Costs & Risk

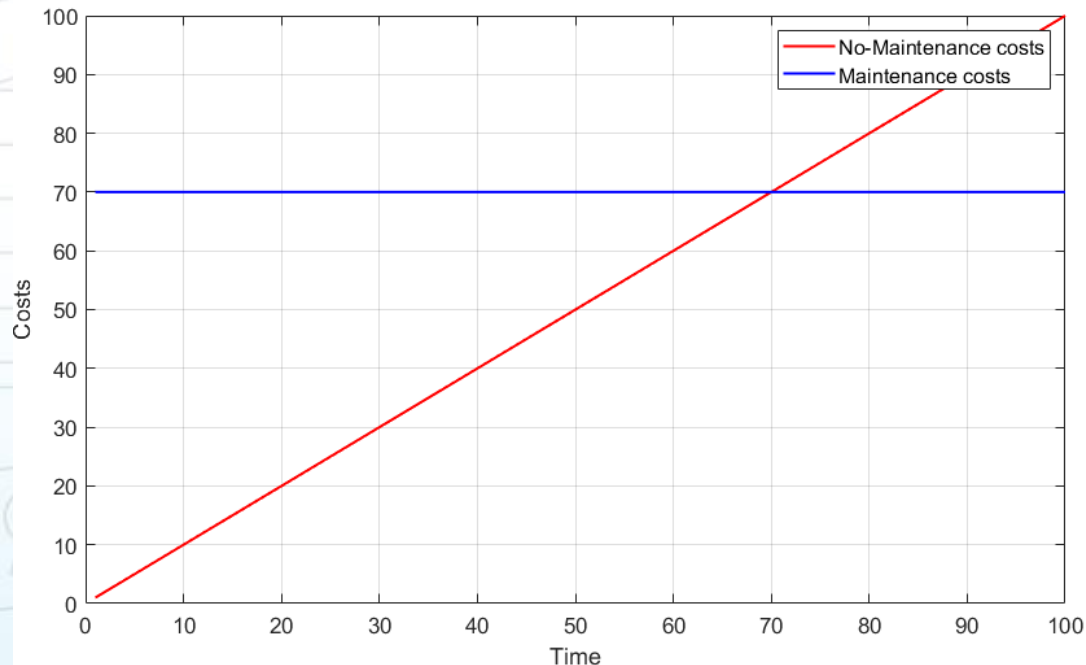
Costs & Risk 1/2

As the curve representing the RUL approaches its threshold value, the probability of failure increases. The choice to carry out a maintenance action must therefore follow two decision logics: the **probability of failure** and the **costs**.



Costs & Risk 2/2

Not carrying out maintenance leads to **lower maintenance costs** but exposes to **higher risks** (and thus failure costs) that increase with operating hours. Conversely, investing a certain amount of capital for a maintenance action entails a **cost** that generates a **lower risk**.



Conclusions

Conclusions

In this study, the possibility of carrying out a degradation assessment of gas turbine systems through a **data-driven** statistical approach was evaluated. The following results are observed:

- The **sensor selection** activity based only on PE led to the selection of the most significant sensors.
- **Multi-objective optimization** achieved satisfactory levels of PE and variance.
- The fused indicator proved capable of predicting **RUL** effectively.

The absence of specific material analysis and geometric modelling (which also requires plant shutdowns) ensures **minimal implementation costs** compared to other approaches.



Thanks for your attention

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