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Introduction

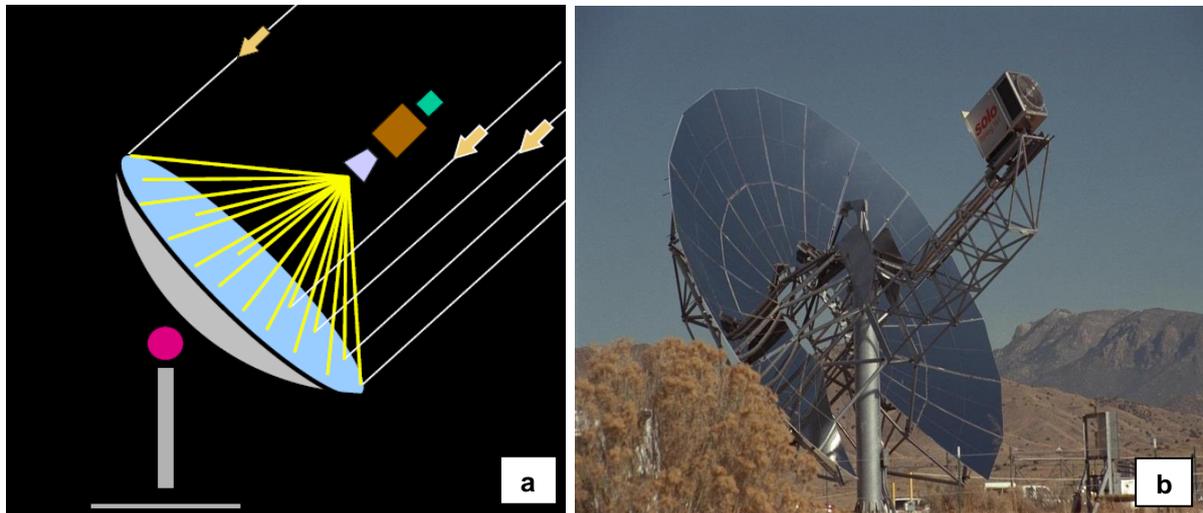
Sun can be seen as a Nuclear Fusion Reactor (NFR), the corresponding energy being delivered into the surrounding as matter and magnetic radiation.

The behavior of the thermal machines is based on thermodynamic cycles that take benefits from the cycle maximum temperature achieved by the working fluid (WF). Thus, the concentration of the solar radiation (energy) is a must to get the technologically possible power from the solar radiation impinging a square meter of reflecting surface.

One of the CSPs that have been studied and developed for terrestrial applications is based on a parabolic dish (Figure 1a) that allows to reach high temperatures concentrating the radiation in a focus.

The Stirling engine, based on volumetric mechanisms, has been the most important candidate to pursue the energy conversion from radiation to heat and electricity. In this way the Dish/Stirling (Figure 1b) has been deeply investigated and it has achieved good thermodynamic performance in comparison with other CSP Systems.

Since the OMSoP Project is aimed at the development of an alternative thermodynamic engine based on turbomachines working according to Brayton Cycle, this report gives the State of the Art of the Dish/Stirling CSP technology, the main achievements and the critical issues.



*Figure 1- a) Schematic view of a Parabolic Dish System;
b) Example of Dish/Stirling System (SES/SOLO Prototype)*

Scientific and Technical Background

The Sun surface temperature is estimated to be around 6000 to 10000 K, the related radiation energy flux being about 63 MW/m². The energy flux that arrives out of the Earth atmosphere is about 1300 W/m², the distance from the Sun being 1.5•10¹¹ m. The Sun diameter is about 1.39•10⁹ m, thus the Sun surface heat flux is about 4.6•10⁴ times than that out of the Earth atmosphere (Figure 2). To achieve high temperature by the solar radiation means that it must be concentrated. A theoretical study [1] shows the relationship between achievable temperatures and concentration factors.

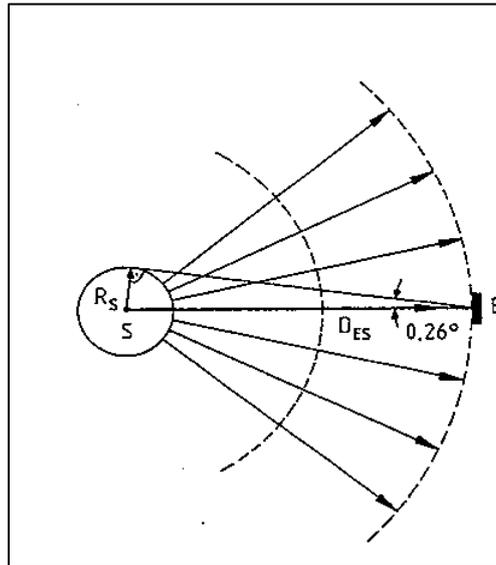


Figure 2 – Sun-Earth geometry. Assuming a Sun spherical surface of radius R_s , E_{sc} is the Solar Constant observed at distance D_{ES} between Earth and Sun [1].

For a given concentrator surface, there are various physical phenomena that bound the power of a solar energy fed engine. One of the most relevant is connected with the re-irradiation from the energy receiver.

A black body reaches the maximum temperature re-irradiating all the received energy, thus no energy can be used to feed an engine.

Other losses are related to:

1. Concentrating systems can utilize only direct irradiation;
2. Optical losses (i.e. imperfect reflection, reflector absorption)
3. Atmospheric losses (e.g. beam attenuation by haze or dust under large focal lengths);
4. Spillage losses (e.g. imperfect tracking and beam aiming under windy conditions);
5. The inability of the reflecting surface to concentrate over a small area (receiver opening) all the intercepted radiation;
6. Thermal re-radiation, conductive and convective losses at receiver/absorber device;

Moreover inoperability due to the irradiance that stays below a certain threshold needed for operation has to be taken into consideration.

Figure 3 represents a typical Sankey diagram for a CSP Sun Energy Capturing System.

Thermal machines to convert heat into work take benefits from the high temperatures of the WF heat injection and from the low temperatures of the WF heat rejection. Also the internal sources of entropy connected with irreversibilities play an important role.

The above can be easily summarized [2,3] by:

1. The Carnot Number $\tau = T_{max}/T_{min}$ that is the ratio between the maximum WF temperature (T_{max}) and the minimum WF temperature (T_{min});
2. The Thermal Sources Concentration Index $\xi = \xi_n/\xi_p$
 ξ_p being the positive thermal source concentration index that takes into account the WF heat weighted averaged temperature of the positive heat sources that warm up the WF. It is the ratio between T_{max} and such a positive heat source average temperature (T_h) (i.e. $\xi_p = T_{max}/T_h$)

ξ_n being the negative thermal source concentration index that is the ratio between T_{min} and the WF heat weighted averaged temperature of the negative heat sources (T_c) that cool the WF (i.e. $\xi_n = T_{min}/T_c$).

3. The Cycle Reversibility Factor $\sigma = \Delta S_n / \Delta S_p$

ΔS_p and ΔS_n being the entropy variations corresponding to the positive addition of heat into the cycle WF and to the negative rejection of the heat from the cycle WF. Such an index, being higher than one takes the cycle irreversibility into account.

The cycle efficiency can be expressed as

$$\eta = 1 - \frac{\sigma}{\tau \xi}$$

T takes the engine technological bounds into account and it is typical of any kind of engine. In particular, for the Stirling engines τ can be up to 3-3.5.

The Concentration Factors ξ_p and ξ_n are improved by the internal heat recuperation and become 1 for iso-temperature both positive WF heat injection and negative WF heat rejection.

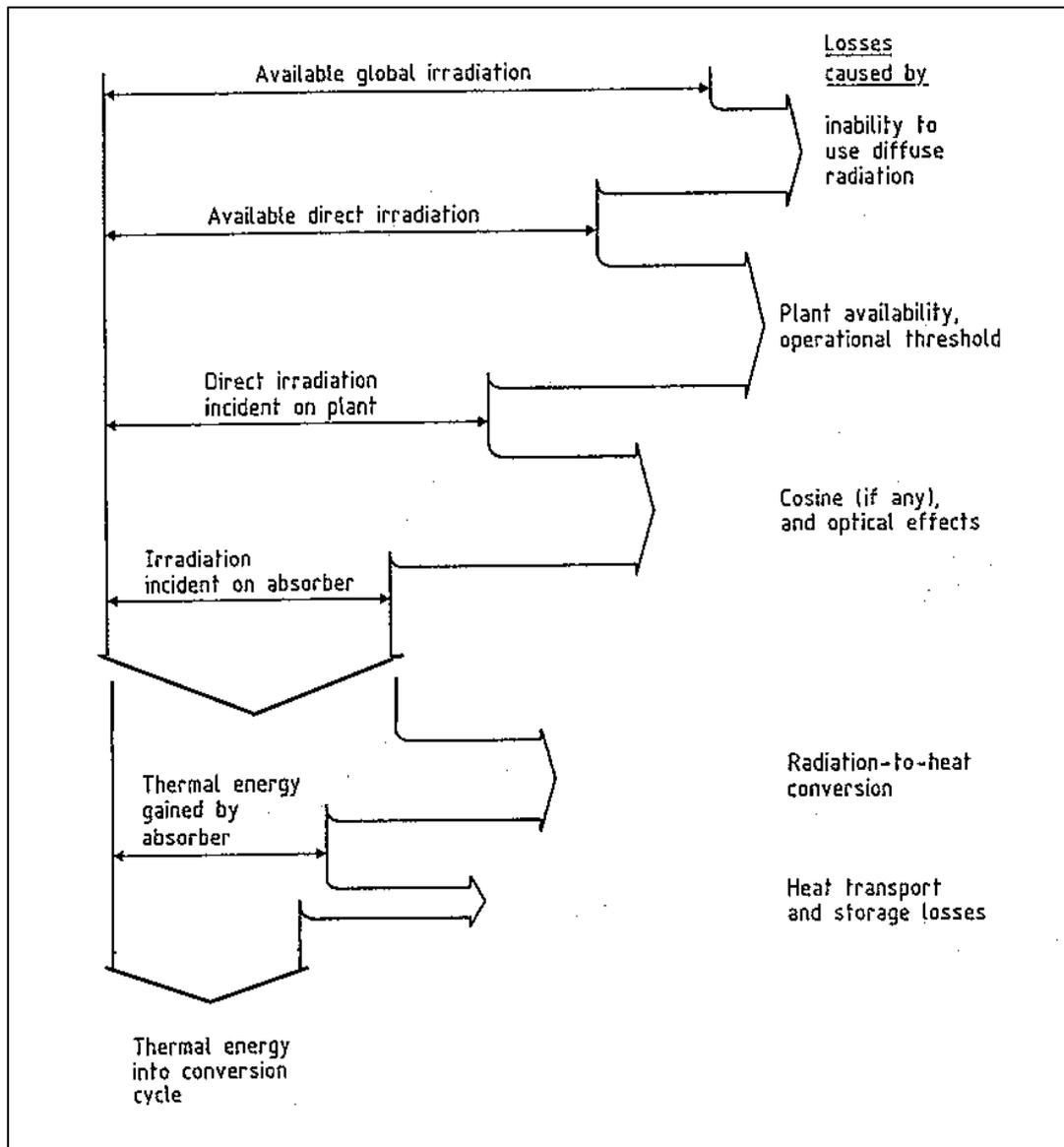


Figure 3 – Simplified Sankey Diagram for the Dish/Receiver System

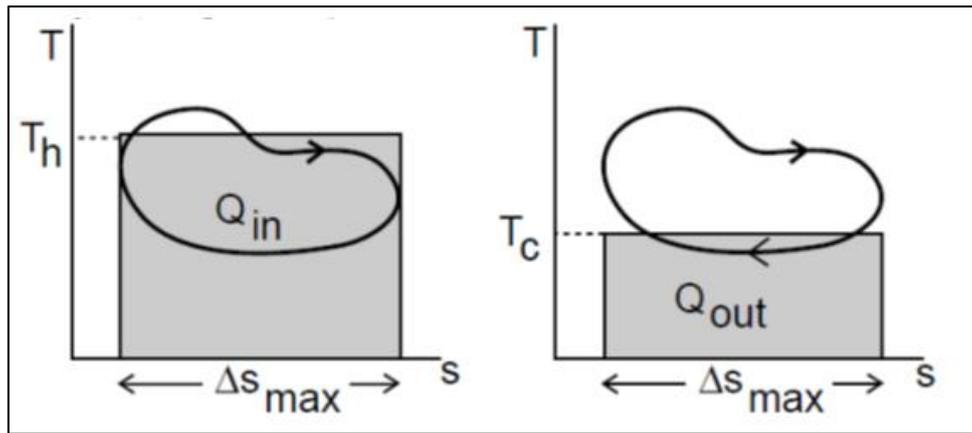


Figure 4 – Temperature-Entropy generic direct cycle representation

In [4] an amply description of the Stirling engine technologies and their applications to solar-powered systems are given.

The engine operation in shown in Figure 5. It consists in:

- an isothermal compression process (1-2): the WF is in the cold space. The power piston, which is approaching the Top Dead Center (TDC), compresses the working fluid from 1 to 2 at constant temperature;
- a constant-volume heating process (2-3): the displacer moves from TDC to the Bottom Dead Center (BDC). There is a WF transfer from the cold space to the hot space, while the power piston remains stationary at the TDC. The displacer pushes the WF into the hot space, passing through a regenerator which has stored heat. Heat given up by the regenerator raises the WF temperature and pressure at constant volume. Heat stored in the regenerator is given to the WF.
- an isothermal expansion process (3-4): after the displacer has pushed all the WF into the hot space, it is then kept at rest at its BDC. The WF is in the hot space and expands at constant temperature. The power piston is pushed from TDC to BDC by the increased pressure giving a mechanical power to a shaft.
- a constant-volume cooling process (4-1): the power piston remains stationary and it is ready to travel back to TDC under flywheel momentum and the sucking action of the partial vacuum created by the falling pressure. The displacer moves from BDC to TDC and transfers the to the cold space, where the pressure decreases and a partial vacuum is created, through the regenerator, decreasing the WF temperature and pressure at constant volume. Heat is transferred from the WF to the regenerator.

As shown in Figure 6 and reported in [4], “three different configurations (alpha, beta, and gamma) are commonly used. Each configuration has the same thermodynamic cycle but different mechanical design characteristics”.

- In the alpha-configuration the displacer is not used. Two pistons (hot and cold pistons) are used on either side of the heater, regenerator, and cooler. These pistons move uniformly in the same direction to provide WF constant-volume heating or cooling processes. When all the WF has been transferred into one cylinder, one piston is fixed and the other piston moves to expand or compress the WF. The expansion work is done by the hot piston while the compression work is done by the cold piston.
- In the beta-configuration, displacer and power piston are in the same cylinder. The displacer moves the WF between the hot and the cold space of the cylinder through heater, regenerator, and cooler. The power piston, located at the cold space of the cylinder, compresses the WF when this one is in the cold space and expands the WF when this one is moved into the hot space.
- The gamma-configuration uses separated cylinders for the displacer and the power piston, with the power cylinder connected to the displacer cylinder. The displacer moves the WF between the hot space and the cold space of the displacer cylinder through the heater, regenerator, and cooler. In this configuration, the power piston both compresses and expands the WF. The gamma-configuration with double-acting piston arrangement has theoretically the highest possible mechanical efficiency.

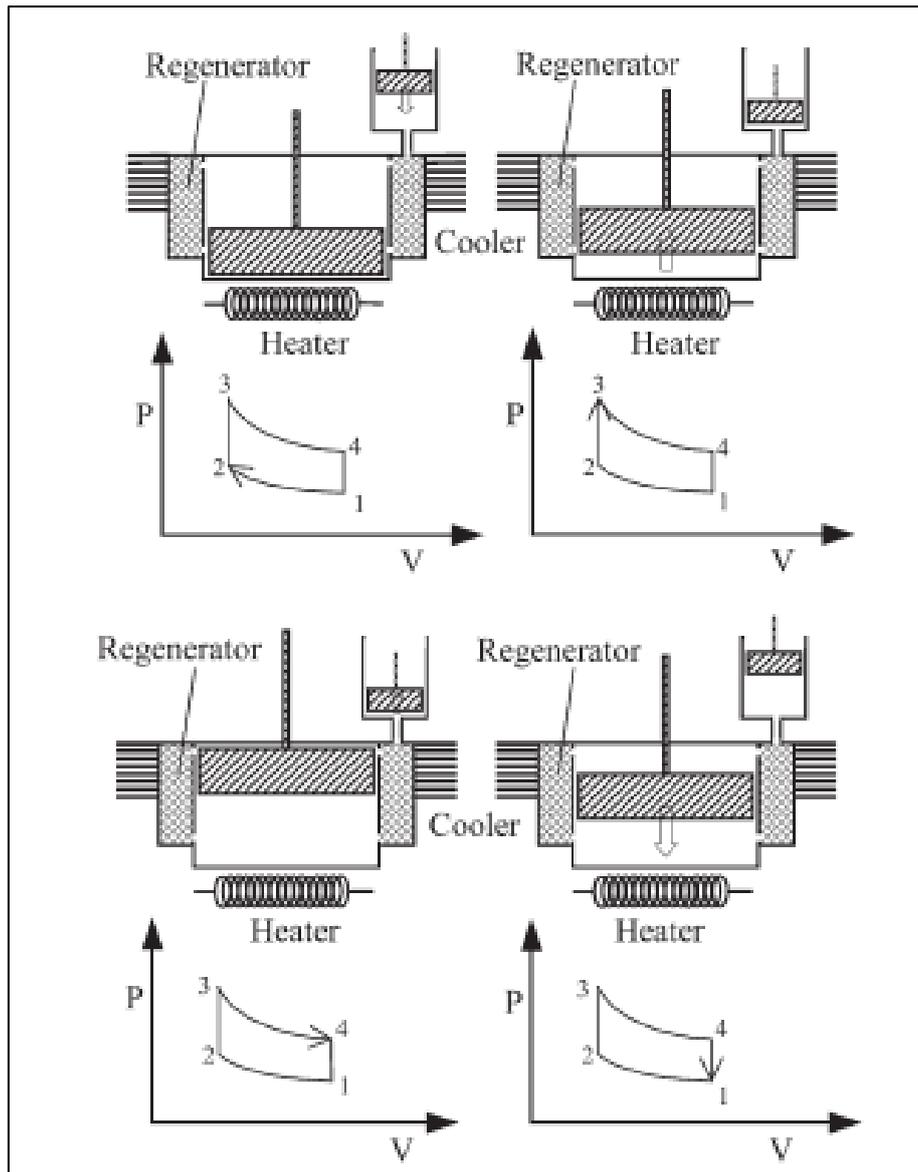


Figure 5 – Stirling engine operation [4]

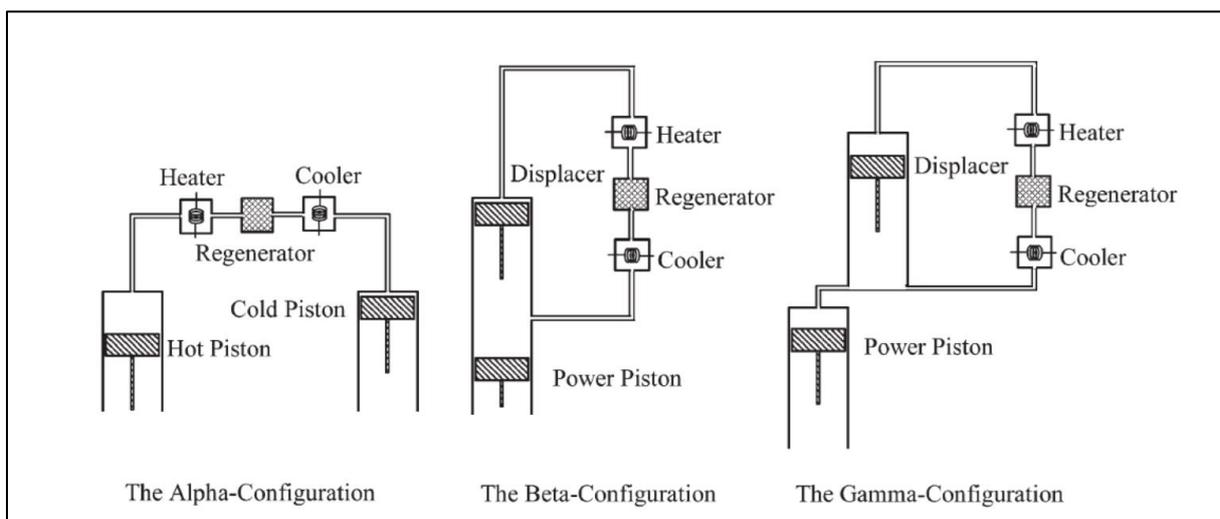


Figure 6 – Mechanical configurations for Stirling engines [4]

In [4] there is also a review of the main formulas for the calculation of the indicated work, power and efficiency of the engine. The growth in interest for the Stirling engine is connected with the development of new materials and the application of WFs like helium and hydrogen which have a low viscosity. It allows the reduction of pumping losses

According to such characteristics, Stirling engine is a good candidate for CSP Systems based on Dishes and it competes with the Brayton cycle based engines that are the subject of the OMSoP Project.

Various developments have been performed and the present Deliverable reports the State of the Art of Dish/Stirling CSP Systems.

Bibliographical Review

The major literature that has been analyzed to extract the information is listed in the last chapter and the main papers are summarized in the following.

The main components of a Dish/Stirling engine Systems are:

1. The solar concentrator (Dish) with the sun-tracking system;
2. The solar receiver;
3. The Stirling engine with the electric generator/alternator.

Solar Concentrator (Dish)

In [5,6] a complete description of such components is given. The solar concentrator is generally a point-focus dish with a parabolic or a quasi-parabolic shape, but there are examples of point-focus concentration systems based on flat mirrors overlapped radiation [7,8].

In Figure 7 some configurations developed for the Dish are shown.



Figure 7 – a)McDonnell Douglas Corporation Concentrator (1984) [6]; b)EnviroDish (2006)[9] c) DOE Faceted Stechde-Membrane Dish, 1992 [6] d) Heliofocus concentrator prototype 2013 [8] e)Dish-Stirling Prototype, Gardenshow in Pforzheim (1989/1991) [27]

In [6] concentrators are divided into three main categories:

1. Glass-Faceted Concentrators: they use spherically curved, individually alignable glass mirror facets mounted on parabolic-shape structures. An example is given in Figure 6a. This design guarantees high concentration ratios but it has not been implemented for Dish/Stirling Farms Projects because it is heavy, expensive and requires accurate alignment of a large number of mirrors. In Figure 8 an example of curved glasses mirror is given;
2. Full-Surface Paraboloid Concentrators: the entire surface forms a paraboloid. Two prototypes have been developed in the past for Dish/Stirling applications, but tests showed a low optical efficiency (about 60%);
3. Stetched-Membrane Concentrators: Membranes are made of thin metal or plastic sheeting with a reflective coating. Such membranes are stretched over both sides of a metal ring. Various designs have been developed. In the Single Faceted-Stretched Membrane Dishes the surface shape approximation to a paraboloid is given by initially forming the membrane and using the pressure difference between front and back to support the shape and maintain its shape. For Multiple Faceted-Stretched Membrane Dishes the same technique is used to obtain parts of the reflector. The main advantage of these solution is the reduction of costs maintaining good performance. Examples are shown in Figure 6 b and e.



Figure 8 - Segmented Dishes (Curved Glasses)[27]

Moreover, in order to have a good radiation concentration, a two-axes tracking system is necessary. In [6] a description of the two ways used to implement this is given.

1. Azimuth-elevation tracking: the Dish rotates in a plane parallel to the earth (azimuth) and in another plane perpendicular to it (elevation). This gives the collector up/down and right/left rotations;
2. Polar tracking method: the collector rotates about an axis parallel to the earth's axis of rotation, while the other rotational axis (declination axis) is perpendicular to the polar axis.

In Figure 9 examples of the two tracking systems are shown.

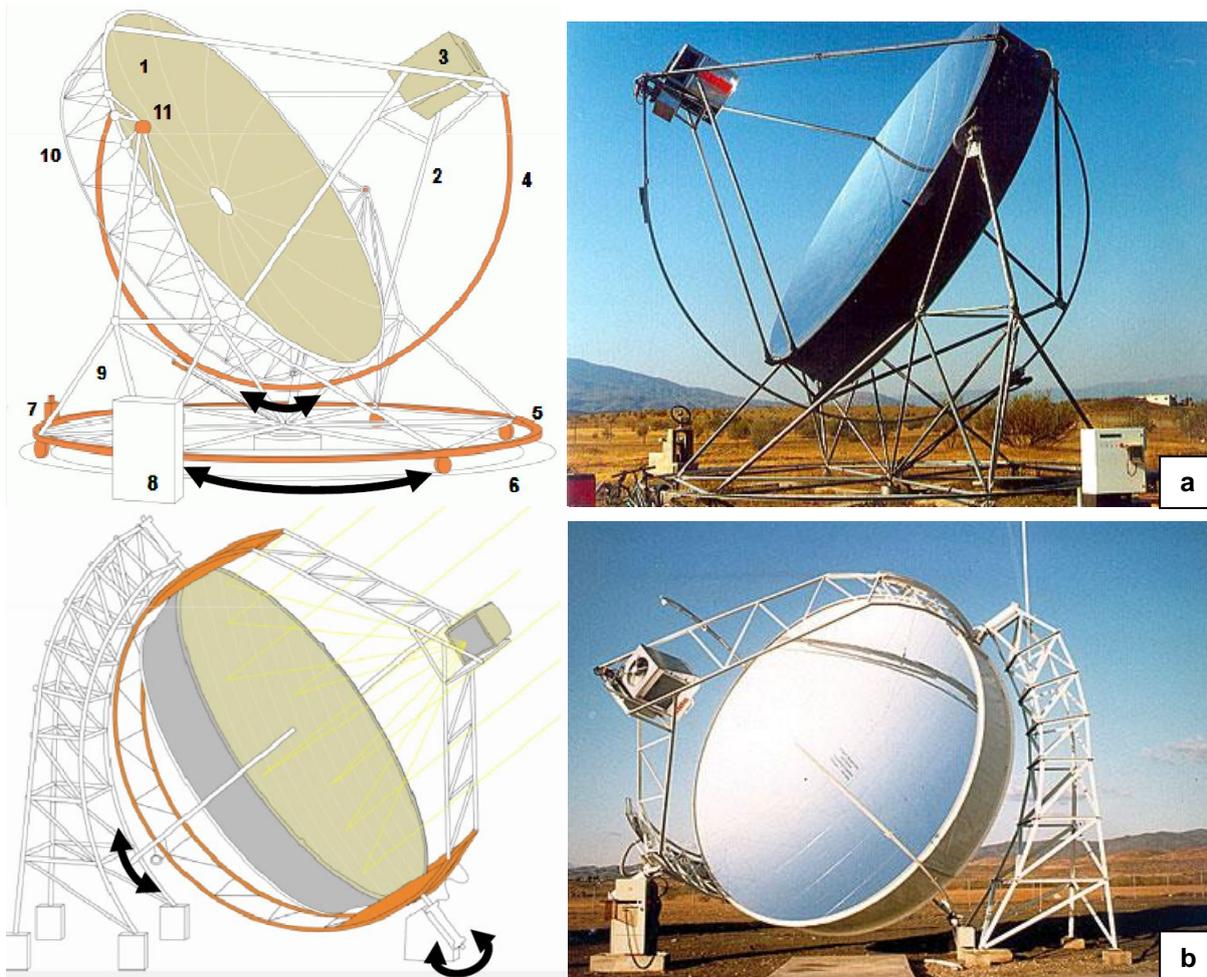


Figure 9 - a) Azimuth elevation Tracking System- Distal II Almeria 1996 b) Polar Tracking System- Distal I Almeria 1992

Solar Receiver

The receiver has to absorb the reflected and concentrated solar radiation and has to transfer the heat to a WF. In [6] there is a description of the concepts applied in the design of receivers for Dish/Stirling Engine Systems.

Such components have a cavity with a small opening through which the concentrated sunlight enters. The receiver aperture geometry has to limit radiation and convection losses. To transfer the absorbed radiation to the WF various methods can be adopted. In particular, directly illuminated tube receivers use small tubes through which the WF flows are placed directly in the concentrated solar flux region. Tubes form the absorber surface. Otherwise other receiver use a liquid-metal intermediate heat-transfer fluid. The liquid metal is vaporized on the absorber surface and condenses on tubes carrying the WF (Reflux Receiver). In Figure 10 the two concepts are illustrated.

In Figure 11 details of modern solar receivers for Solar Stirling engines are shown.

Stirling Engines

Stirling engines used for Dish Solar Systems are derived from aerospace applications. In 70s and 80s various concepts have been developed and proposed for terrestrial purposes. In [10] a review of such designs is given and some technologies are illustrated in Figures 12,13 and 14. Solar Stirling engines generally operate at a maximum temperature of 650-800°C according to material thermal limits, with an engine conversion efficiency of around 30-40%.

Usually helium and hydrogen are used as WF: helium has fewer material compatibility problems than hydrogen and it is safer but on the thermodynamic point of view, hydrogen allows a higher engine efficiency value.

Dish/Stirling Systems are equipped with two kind of engines:

1. Kinematic engines;
2. Free-piston engines.

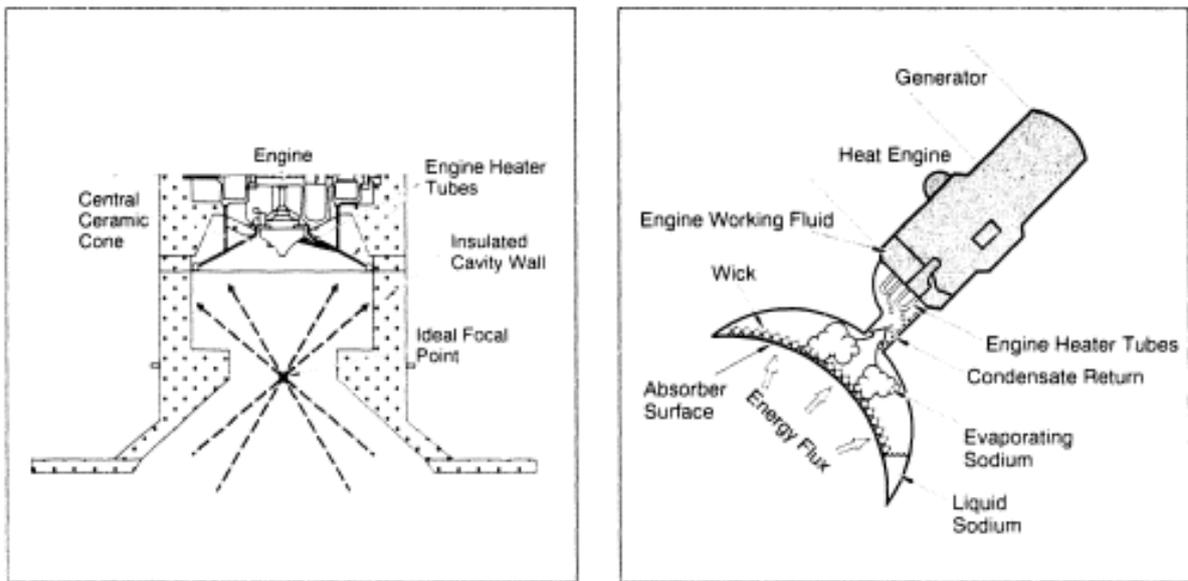


Figure 10 – a) Directly illuminated Tube Receiver b) Reflux heat-pipe Receiver

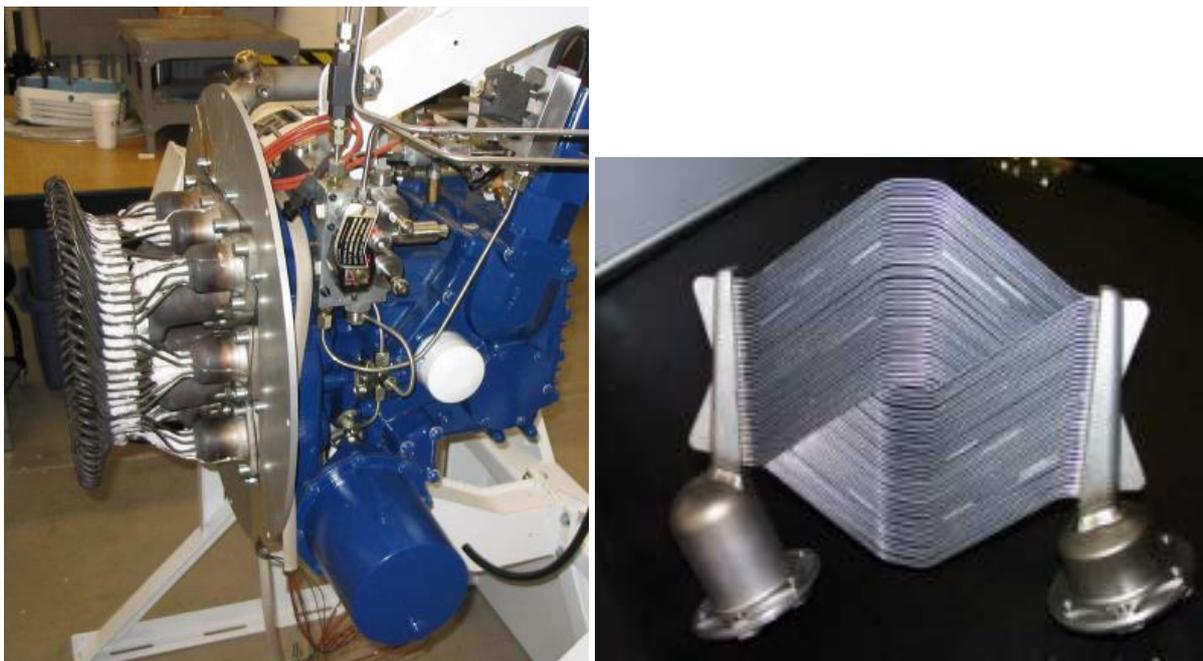


Figure 11 - Details of modern solar receivers for Solar Stirling engines

In a kinematic engines the power piston is mechanically connected to a rotating output shaft (Figures 15 and 17). As shown in Figure 16 for the STM 4-piston engine, some prototypes have been designed also in a hybrid configuration (they are equipped with a combustor) [11,18].

The free-piston engine is based on another concept. It is not mechanically connected to an output shaft, but bounces alternately between the space containing the WF and a spring that usually contains a gas. Dynamics of the spring/mass systems determines piston frequency. A magnet is attached to the power piston and electric power is generated as it moves past stationary coils. In Figure 18 a scheme of the INFINIA free-piston engine is shown [12].

In [6] are reported design and performance specifications for the main components of Dish/Stirling Systems prototype developed and tested before 1994. The table is shown in Figure 19.

In the next chapter information collected for prototype and commercial modules are illustrated.

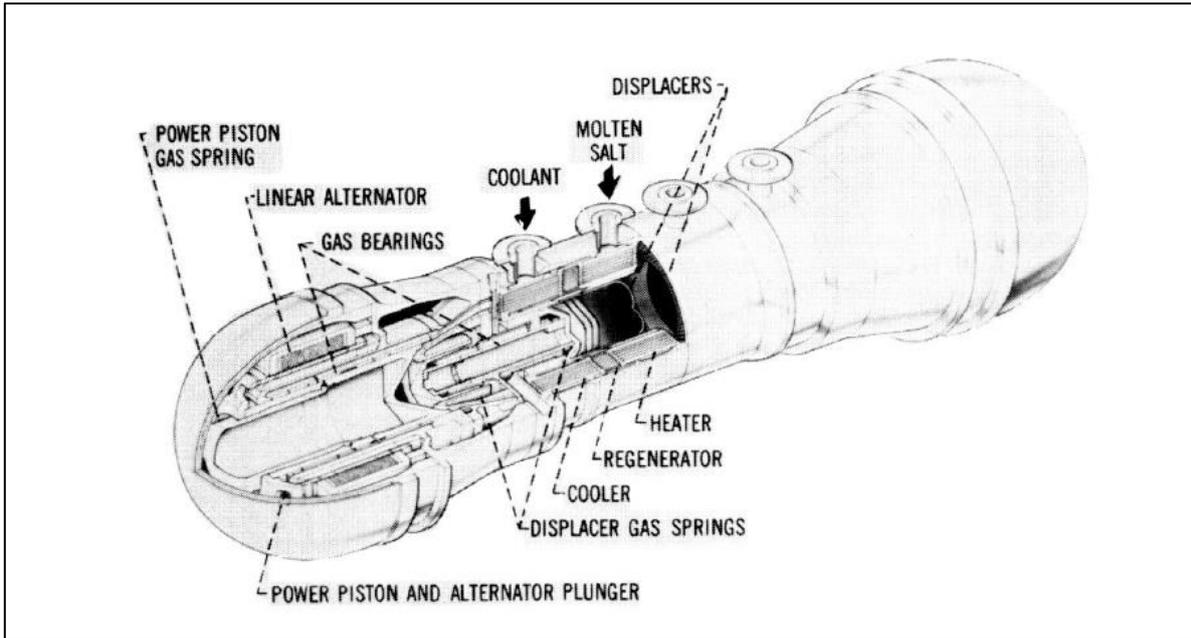


Figure 12 - 25 kWe Space Power Demonstrator Engine

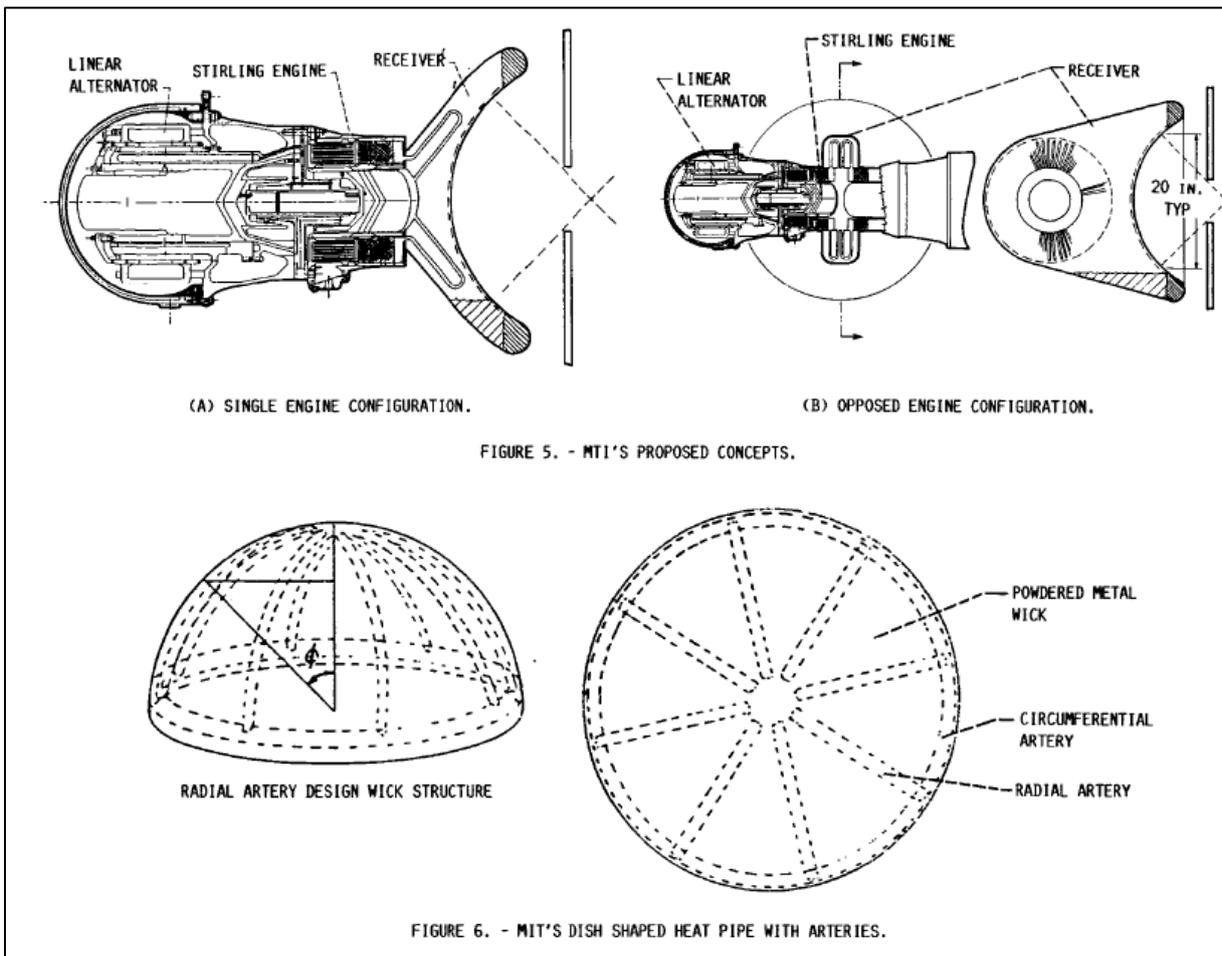


Figure 13- MIT's preliminary Dish-Stirling engine

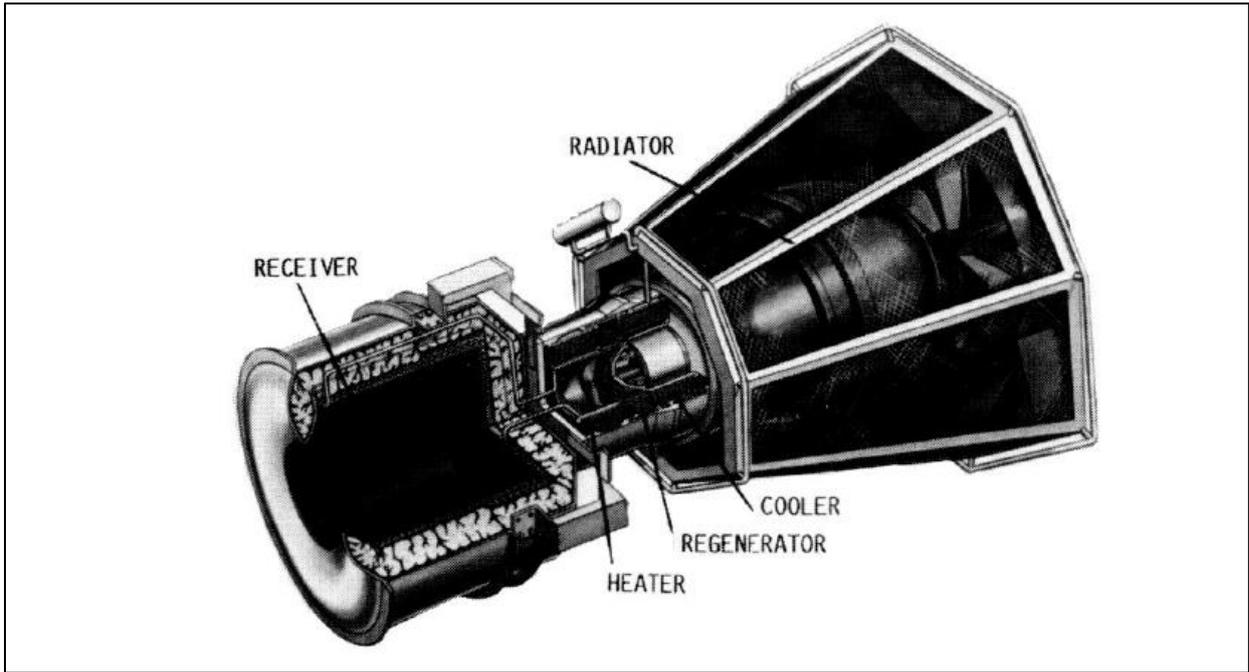


Figure 14 – Conceptualized Free-Piston Stirling engine conversion system

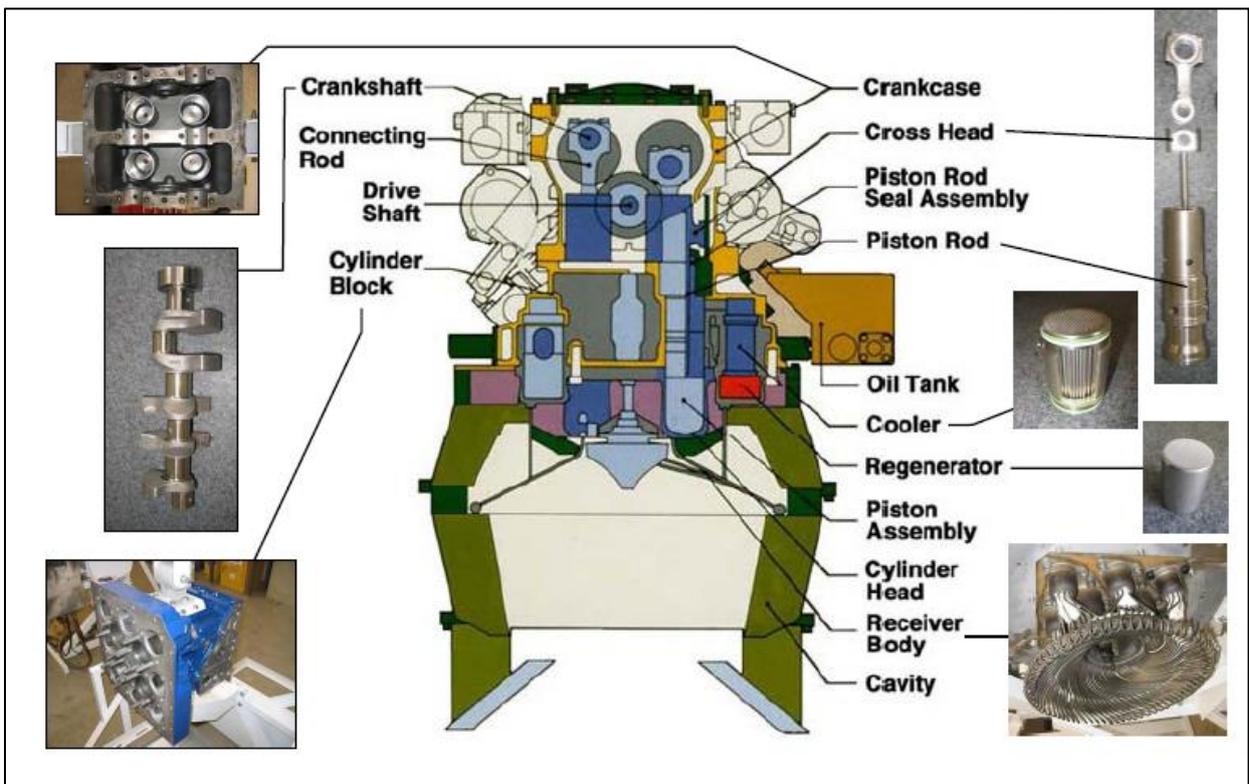
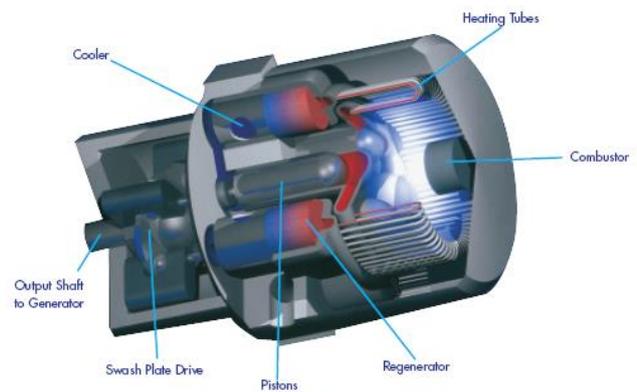
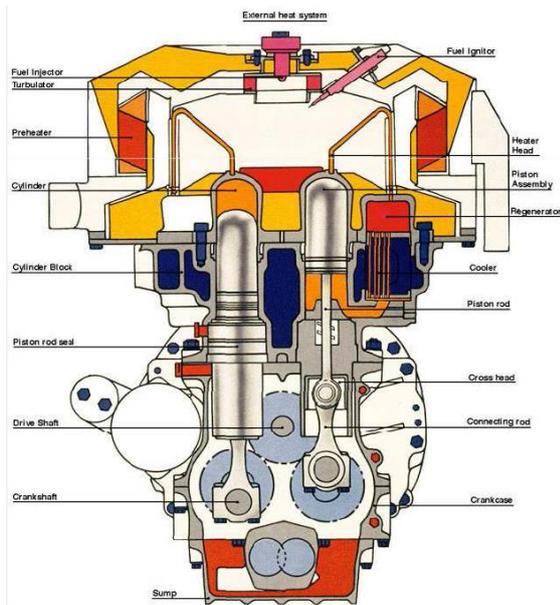


Figure 15 – Scheme of a kinematic engine and its main components



STM 4-Piston Stirling Engine

Figure 16 - 4-Piston hybrid kinematic Solar Stirling engine

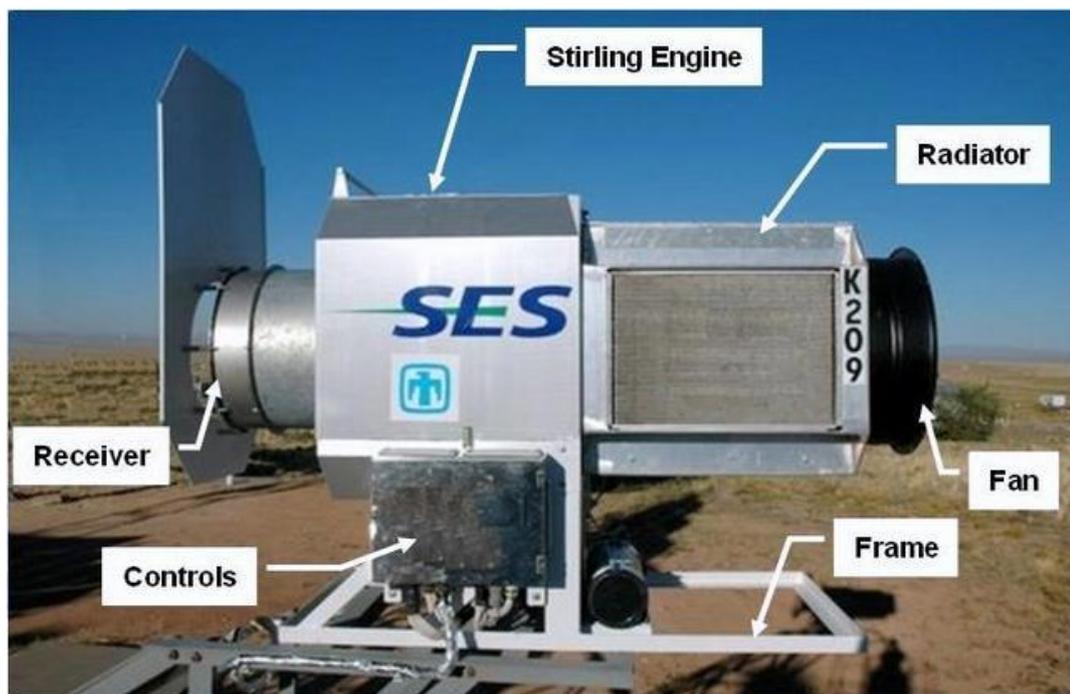


Figure 17- SES kinematic Solar Stirling engine. It is a 4 cylinder, each with a 95cc displacement engine (4-95 engine) that evolved from the Philips engines of the 1960's

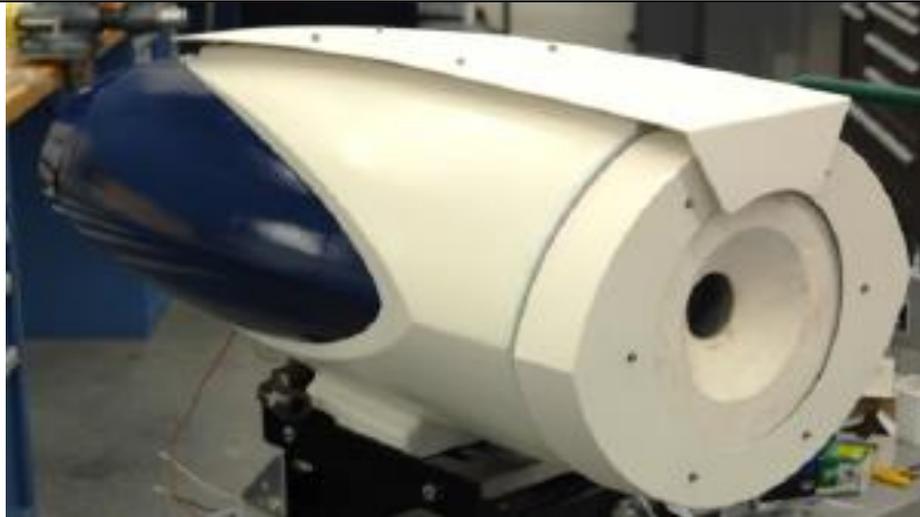
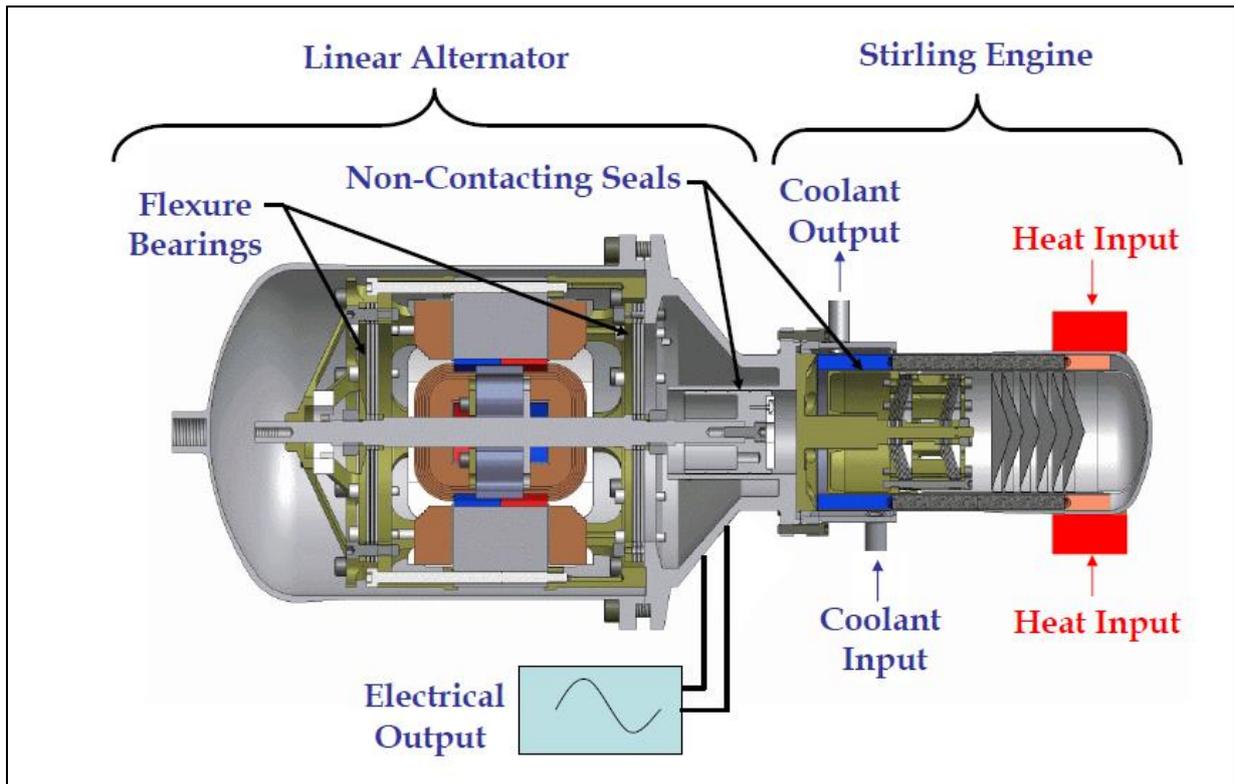


Figure 18 - INFINIA Free Piston Engine: scheme and picture [12]

SYSTEM

Name	Vanguard	MDAC	German/Saudi	SBP 7.5-m	CPG 7.5-kW	Aisin/Miyako	STM Solar PCS
Year	1984	1984-88	1984-88	1991-	1992-	1992-	1993-
Net Electricity*	25 kW	25 kW	52.5 kW	9 kW	7.5 kW @ 950 W/m ²	8.5 kW @ 900 W/m ²	25 kW (design)
Efficiency ⁺	29.4% @ 760° C gas temp.	29% - 30%	23.1%	20.3%	19% @ 950 W/m ²	16 % @ 900 W/m ²	— ⁺⁺⁺
Number	1	6	2	5	3 built, 14 planned	3 planned	1
Location (no.)	CA	CA (4), GA, NV	Riyadh, Saudi Arabia (2)	Spain (3) Germany (2)	CA, TX, PA,	Miyako Is, Japan	SNL-TBC
Status	Testing completed	Testing completed	Occasional ops.	Testing now	Initial testing of 5-kW prototype	Fabrication	—

CONCENTRATOR

Manufacturer	Avanco	MDAC	SBP	SBP	CPG	CPG	—
Diameter ^{**}	10.57 m	10.57 m	17 m	7.5 m	7.3 m	7.5 m	—
Type	Faceted glass mirrors	Faceted glass mirrors	Stretched membrane	Stretched membrane	Stretched membrane	Stretched membrane	—
No. of Facets	336	82	1	1	24	24 ^{***}	—
Size of Facets	0.451x0.603 m	0.91x1.22 m	17 m dia.	7.5 m dia.	1.524 m dia.	1.524 m dia.	—
Surface	Glass/silver	Glass/silver	Glass/silver on stainless steel	Glass/silver on stainless steel	Aluminized plastic film	Aluminized plastic film	—
Reflectance (initial)	93.5%	91%	92%	94%	85% to 78%	85% to 78%	—
Concentration ⁺	2750	2800	600	4000	1670	1540	—
Tracking	Exocentric gimbal	Az-el	Az-el	Polar	Polar	Polar	—
Efficiency	89%	88.1%	78.7%	82%	78%	78%	—

ENGINE

Manufacturer	USAB	USAB	USAB	SPS/Solo	Sunpower/ CPG	Aisin Seiki	STM / DDC
Model	4-95 Mk II	4-95 Mk II	4-275	V-160	9-kW	NS30A	4-120
Type	Kinematic	Kinematic	Kinematic	Kinematic	Free-piston	Kinematic	Kinematic
Power (elect.)	25 kW	25 kW	50 kW	9 kW	9 kW	30 kW (derated to 8.5 kW) ^{****}	25 kW
Working Gas	Hydrogen	Hydrogen	Hydrogen	Helium	Helium	Helium	Helium
Pressure (max.)	20 MPa	20 MPa	15 MPa	15 MPa	4 MPa	14.5 MPa	12 MPa
Gas Temp. (high)	720°C	720 C	620°C	630 C	629°C	683 C	720°C
Peak Efficiency	41%	38% - 42%	42%	30%	33% ⁺⁺	25%	42%

RECEIVER

Type	Direct tube irradiation	Direct tube irradiation	Direct tube irradiation	Direct tube irradiation	Sodium heat pipe	Direct tube irradiation	Direct tube irradiation
Aperture Diameter	20 cm	20 cm	70 cm	12 cm	18 cm	18.5 cm	22 cm
Peak Flux	75 W/cm ²	78 W/cm ²	50 W/cm ²	80 W/cm ²	30 W/cm ²	30 W/cm ²	75 W/cm ²
Tube Temp. (max.)	810°C	—	800°C	850 C	675°C ^{****}	780 C	800°C
Efficiency	90%	90%	80%	86%	86%	65%	85% to 90%

- * At 1000 W/m² unless otherwise noted
- ** Equivalent disk
- *** 32 for temporary high output
- + Geometric concentration ratio, defined in Chapter 3
- ** Includes alternator
- +++ Depends on concentrator used
- **** Heat pipe internal temperature (Na vapor)

Figure 19 - Specifications of Dish/Stirling prototype systems(till 1994)[6]

Developed Dish/Stirling Systems

Prototypes

As reported in [13], Modern Stirling engines coupled with directly-illuminated solar receivers were developed in the late 1970s and early 1980s by United Stirling AB, Advanco Corporation, McDonnell Douglas Aerospace Corporation (MDA), NASA's Jet Propulsion Laboratory, and DOE. They were initially for aerospace purposes, but, after a few years, they were applied also for terrestrial electric power generation with Dish/Stirling engine Systems.

In 1984, the Advanco Vanguard 25 kW prototype recorded a solar-to-electric conversion efficiency of 29.4% (net) using the United Stirling PCU. This efficiency is defined as the net electrical power delivered to the grid, taking into account the electrical power needed for parasitics, divided by the direct normal insolation incident on the mirrors. In Figure 20 the prototype is shown.

MDA subsequently attempted to commercialize a system using the United Stirling PCU and a dish of their own design. Eight prototype systems were produced by MDA before the program was canceled in 1986 and the rights sold to Southern California Edison (SCE). The cancellation of the dish/Stirling program was part of MDA's decision to cancel all of their energy related activities, despite the excellent technical success of their dish/Stirling system. The MDA systems routinely converted sunlight incident on the concentrator's mirrors to electricity with net efficiencies of about 30%. Southern California Edison Company continued to test the MDA system on a daily basis from 1986 through 1988. During its last year of operation, it achieved an annual efficiency of about 12%, including system outages and all other effects such as mirror soiling, and an annual efficiency of over 23% was determined to be achievable excluding outages.

In the early 1990s, Cummins Engine Company attempted to commercialize Dish/Stirling systems based on free-piston Stirling engine technology. The Cummins development efforts were supported by SunLab through two 50/50 cost shared contracts. The Dish/Stirling Joint Venture Program (DSJVP) was started in 1991 and was intended to develop a 5 to 10 kW dish/Stirling system for remote power applications. The Utility Scale Joint Venture Program (USJVP) was started in late 1993 with the goal of developing a 25 kWe dish/engine system for utility applications. However, largely because of a corporate decision to focus on its core diesel-engine business, Cummins canceled their solar development in 1996. Technical difficulties with Cummins' free-piston Stirling engines were never resolved.

In 1993, another USJVP contract was initiated with Science Applications International Corporation (SAIC) and Stirling Thermal Motors (STM) to develop a dish/Stirling system for utility-scale applications. The SAIC/STM team successfully demonstrated a 20-kW unit in Golden, Colorado, in Phase 1. In December 1996, Arizona Public Service Company (APS) partnered with SAIC and STM to build and demonstrate the next five prototype dish/engine systems in the 1997-1998 time frame. SAIC and Stirling Thermal Motors, Inc. (STM) worked on next-generation hardware including a third-generation version of the STM 4-120, a faceted stretched-membrane dish with a face-down-stow capability, and a directly-illuminated hybrid receiver. The overall objective is to reduce costs while maintaining demonstrated performance levels. Phase 3 of the USJVP calls for the deployment of 1 MW Dish/engine systems in a utility environment.

Then, other developers continued to develop such systems. For example, Stirling Energy Systems (SES) purchased the rights of the MDA technology and improved the system recording a conversion efficiency up to 31% (Figure 21) [9].

In Figure 21 a Table from [13] with the major characteristics of some 80s-90s prototype is presented. They range in size from 5 to 50 kWe and they have been built in United States, Germany, Japan and Russia.

In 2000s other prototypes of interest have been developed and installed in addition to SES systems. The EnviroDish System installed at Odeillo – Spain in 2006 is just an example. It has been developed on the basis of the experience of Distal I and II prototypes and the EuroDish System. The Envirodish used a SOLO V161 Stirling engine with hydrogen as WF and it recorded a peak efficiency of 24%.

Another relevant example is the INFINIA 3,2 kWe Dish/Stirling System. It was marketed into 2006 and in 2010 about 100 systems have been installed in the world. The Company declares an engine efficiency of 32%, 86% of net reflectivity for the concentrator and a net efficiency of the system of 24% after inverter and parasitic losses [12] and, launches its product as the best commercial module comparing test results (Figure 23) [14]. On the basis of the 3.2 kWe Module, Infinia started the development of a 30 kW 6-cylinder configuration (a scheme is shown in Figure 24) based on the existing module [14]. The project was also funded by the Department of Energy (DOE) of the USA in

order to approach the target of 7-10 c\$/kWh by 2015 and 5-7 c\$/kWh by 2020 using economies of module size and high quantity production. After the 2008 announcement, no other information related to this project are available.

Since the WF has to be cooled, the Dish/Stirling System can have an interesting utilization where cogeneration is required. A commercial module available in the last years on the Italian market is the INNOVA Trinum: a Dish/Stirling system that provides 1 kWe and 3kWth. For such a system INNOVA declared an overall conversion efficiency of 55.2%, an electrical efficiency of 13.8% and a thermal efficiency of 41.4% based on 725 W/m² DNI (Figure 25) [15].



Figure 20 –ADVANCO Vanguard Dish/Stirling prototype system



Figure 21 - SES (Sandia Lab, Albuquerque, 2005)

Company	Concent Type	Optical Surface	Dish Area (m ²)	Rec Type	Total Hours/ (Nos. of systems)	Engine Company	Engine Type	Rated System Output (kW _e)	Steady, system Output (kW _e)	Steady, system Effic (%)
Avanco Vanguard System	Faceted Glass-Metal	Glass	86.7	DIR	1996 (1)	United Stirling	Double-Acting 4-cyl, KSE	25	~ 20 - 22	23
Schlairch Bergermann und Partner	Stretched Membrane	Glass	227	DIR	NA (2)	United Stirling	Double-Acting 4-cyl, KSE	50	NA	NA
McDonnell-Douglas SES System	Faceted Glass-Metal	Glass	91	DIR	13,852 (7)	United Stirling	Double-Acting 4-cyl, KSE	25	20 - 22	23
Schlaich Bergermann und Partner	Stretched Membrane Parabolic	thin glass	44	DIR	> 27,000 (6)	SOLO Kleinmotoren	2 cylinder KSE	10	~9.0	~ 18.0
Cummins Power Generation	Faceted Stretched-Membrane	aluminum polymer	44	Heat Pipe	> 1000 (4)	Clever Fellows	2 cylinder FPSE	7	5.2	15
Cummins Power Generation	parabolic gore	thin glass	145	Heat Pipe	~ 25 (1)	Aisen Seiki	4 cylinder KSE	25	~ 20	NA
Science Applications Int. Corp.	Faceted Stretched-Membrane	thin glass	107	DIR ⁺	400 (1)	Stirling Thermal Motors	4 cylinder KSE	20	18.5	22.4

Figure 22 – Details and performance of some Dish/Stirling prototypes

	10 watt (?)	55 watt (NASA)	350/450 watt (comm'l)	1 kW (comm'l)	3 kW (comm'l) Solar	Total
Units	10	30	20	84	2	146
Longest test	100,000	35,000	30,000	25,000	300	N/A
Cumulative test time at Infinia	160,000	30,000	24,000	40,000	500	254,500
Cumulative test time at Clients	90,000	120,000	42,000	30,000	n/a	282,000
Total Test Time	250,000	150,000	66,000	70,000	500	536,500

Figure 23 –Details and tests of various Stirling engine

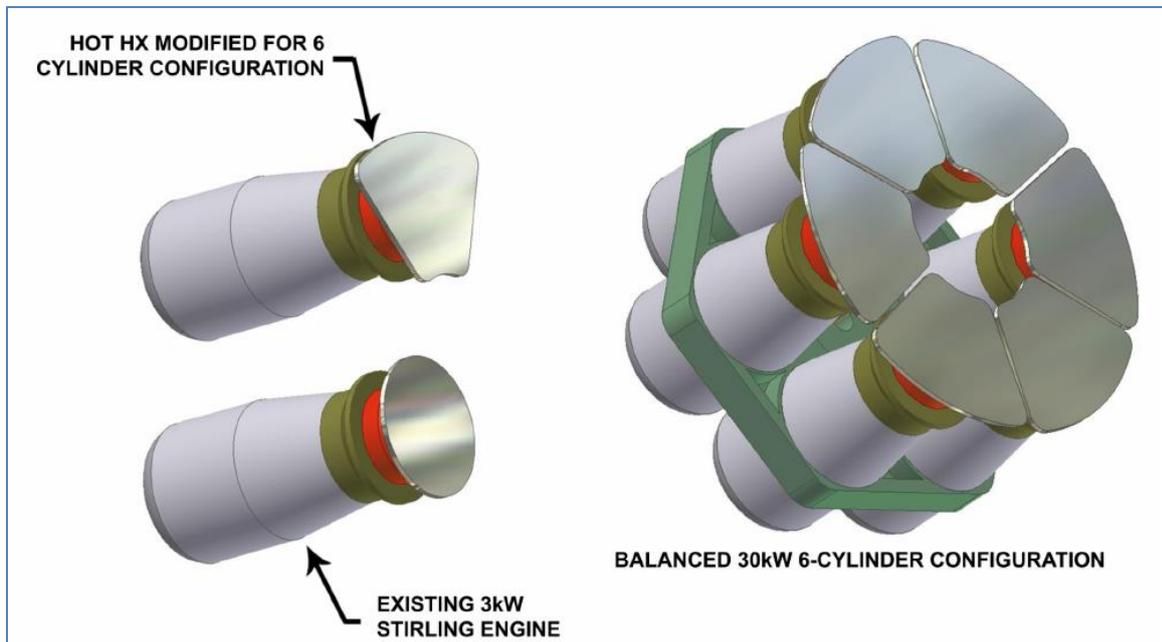


Figure 24 – Infinia 30 kWe 6-cilinder project



Figure 25 – INNOVA Cogeneration Dish/Stirling System Trinum

Concerning Dish/Stirling System costs and performance in [5] the table shown in Figure 26 is reported. System and component costs are from industry sources and independent SunLab analyses. The installed costs include the cost of manufacturing the concentrator and power conversion unit (PCU), shipment to the site, site preparation, installation of the concentrator and PCU, balance of plant (connection to utility grid). The component costs include a 30% profit.

Since costs are extremely sensitive to production rates and the installed costs are extremely dependent on the market penetration, it is not easy to give a “realistic” estimation. Moreover component costs are a strong function of production rates that have to be assumed. The economic life of a dish/engine power plant is 30 years.

As written in [[Solar Dish Engine], costs reported in 1980s prototype was based on the 25 kW dish-Stirling system developed by McDonnell Douglas (MDA) in the mid of 80s but they were in line with the STM 4-120 Stirling engine. Operation and maintenance (O&M) costs are of the prototype demonstration.

Costs reported for the 2000 took significant availability improvements into account, in particular resulting in an annual efficiency of 23% and assuming Stirling engines commercialized for other applications and spare parts and a dedicated staff available.

In addition a modest reduction in the cost of the dish concentrator simply was assumed.

On the basis of a production of 100 modules, resulting in installed costs were nearly \$5,700/kW .

Performance for 2005 was largely based on one of the solarizable engines being commercialized for a non-solar application. A production rate of 2,000 modules per year was assumed.

Performance for years 2010 and beyond was based on an engine performance improvements of well over 10%. A production rate of 30,000 modules per year was assumed.

By 2010 dish/engine technology was assumed to be approaching maturity. So, in this scenario, a typical plant may include several hundred to over a thousand systems. In the table, a typical plant is assumed to be 30 MW.

For the higher production of 2020-2030, 50000 - 60000 modules per year were assumed. Evolutionary improvements in mirror, receiver, and/or engine designs were assumed.

Table 1. Performance and cost indicators.

INDICATOR NAME	UNITS	1980's Prototype		Hybrid System 2000		Commercial Engine		Heat Pipe Receiver		Higher Production 2020		Higher Production 2030	
		1997	+/-%	2000	+/-%	2005	+/-%	2010	+/-%	2020	+/-%	2030	+/-%
Typical Plant Size, MW	MW	0.025	+/-%	1	50	30	50	30	50	30	50	30	50
Performance													
Capacity Factor	%	12.4		50.0		50.0		50.0		50.0		50.0	
Solar Fraction	%	100		50		50		50		50		50	
Dish module rating	kW	25.0		25.0		25.0		27.5		27.5		27.5	
Per Dish Power Production	MWh/yr/dish	27.4		109.6		109.6		120.6		120.6		120.6	
Capital Cost													
Concentrator	\$/kW	4,200	15	2,800	15	1,550	15	500	15	400	15	300	15
Receiver		200	15	120	15	80	15	90	15	80	15	70	15
Hybrid		----		500	30	400	30	325	30	270	30	250	30
Engine		5,500	15	800	20	260	25	100	25	90	25	90	25
Generator		60	15	50	15	45	15	40	15	40	15	40	15
Cooling System		70	15	65	15	40	15	30	15	30	15	30	15
Electrical		50	15	45	15	35	15	25	15	25	15	25	15
Balance of Plant		500	15	425	15	300	15	250	15	240	15	240	15
Subtotal (A)		10,580		4,805		2,710		1,360		1,175		1,045	
General Plant Facilities (B)		220	15	190	15	150	15	125	15	110	15	110	15
Engineering Fee, 0.1*(A+B)		1,080		500		286		149		128		115	
Project/Process Contingency		0		0		0		0		0		0	
Total Plant Cost		11,880		5,495		3,146		1,634		1,413		1,270	
Prepaid Royalties		0		0		0		0		0		0	
Init Cat & Chem. Inventory		120	15	60	15	12	15	6	15	6	15	6	15
Startup Costs		350	15	70	15	35	15	20	15	18	15	18	15
Other		0		0		0		0		0		0	
Inventory Capital		200	15	40	15	12	15	4	15	4	15	4	15
Land, @\$16,250/ha		26		26		26		26		26		26	
Subtotal		696		196		85		56		54		54	
Total Capital Requirement		12,576		5,691		3,231		1,690		1,467		1,324	
Total Capital Req. w/o Hybrid		12,576		5,191		2,831		1,365		1,197		1,074	
Operation and Maintenance Cost													
Labor	¢/kWh	12.00	15	2.10	25	1.20	25	0.60	25	0.55	25	0.55	25
Material	¢/kWh	9.00	15	1.60	25	1.10	25	0.50	25	0.50	25	0.50	25
Total	¢/kWh	21.00		3.70		2.30		1.10		1.05		1.05	

Notes:

1. The columns for "+/-%" refer to the uncertainty associated with a given estimate.
2. The construction period is assumed to be <1year for a MW scale system.

Figure 26 – Evaluation of costs for Dish/Stirling engines

Dish/Stirling Plants

In the last decay some Dish/Stirling Solar Plants have been built on the basis of SES and INFINIA experience. [9]

- **Maricopa Solar Plant**
In January 2010 Tessera inaugurated the Maricopa Solar Plant near Peoria (Arizona) [16]. It was an ambitious 1.5 MWe Plant based on 60 SES Suncatcher Dish/Stirling 25 kWe Systems (with SOLO Stirling engines) (Figure 27). The plant was decommissioned in September 2011 and no other information or details are available for such a plant.
- **Renovalia – Villarobledo**
In 2009 Renovalia started the installation in Villarobledo of a 1 MWe Solar Power Plant based on 3 kWe INFINIA Dish/Stirling engine Modules (Figure 28) [9];
- **PowerPlay Solar – GH Daily**
In 2010 PowerPlay Solar installed a cluster of 30 INFINIA Dish/Stirling Modules at YUMA (Arizona) (Figure 29),
- **Deployment Plants**
In 2007 SES declared that two major contracts were under development for up to 1750 MW of peak capacity, both of them in South California [17].
The first one with Southern California Edison (SCE) for a total capacity of 500 MW with expansion option to 850 MW that means 20000 or 34000 Dish/Stirling Modules.
The second one with San Diego Gas & Electric (SDG&E), sited in the Imperial Valley near El Centro, for a total capacity of 300 MW with expansion option to 900 MW, that means 12000 up to 36000Dish/Stirling Modules.
The construction was scheduled for 2009 for the first plant and for 2008 for the second one but the deployment has never started in both cases. Moreover SES and Tessera have downfallen. A political and economic view of what happened and why is reported in [19].



Figure 27 - Maricopa Solar Power Plant



Figure 28 - Renovalia - Villarobledo



Figure 29 - PowerPlay Solar – GH Dairy Plant

Conclusions

The Dish/Stirling System shows some undeniable advantages: an high efficiency (up to 32% in some cases) also when the insolation is low (Figure 30) [18], a highly modular design, the low requirement of water in comparison with other CSP technologies. But some drawbacks can be highlighted.

In [19] it is pointed out that Dish/Stirling Systems run without a storage system, so they deliver power as a PV and, at the moment, PV technology is cheaper and more reliable. "From an engineering point of view, there are some challenges to work around".

In [20] some of them has been stressed. High efficiency Stirling engine use hydrogen or helium as WF, so the engine piston seals are a critical part. The Eurodish V161 Stirling engine presented a significant gas pressure drop in the motor due to seals leakage after a few days as shown in Figure 31. The WF refilling and the piston seals replacement are, nowadays, a critical point to have a reliable system. Also the pistons (in particular the cooler ones) are a critical part of the Stirling engine because they can rapidly be damaged.

So, although the promising expectations, the reliability of the Dish/Stirling System has to be improved before consider its "real" commercial application.

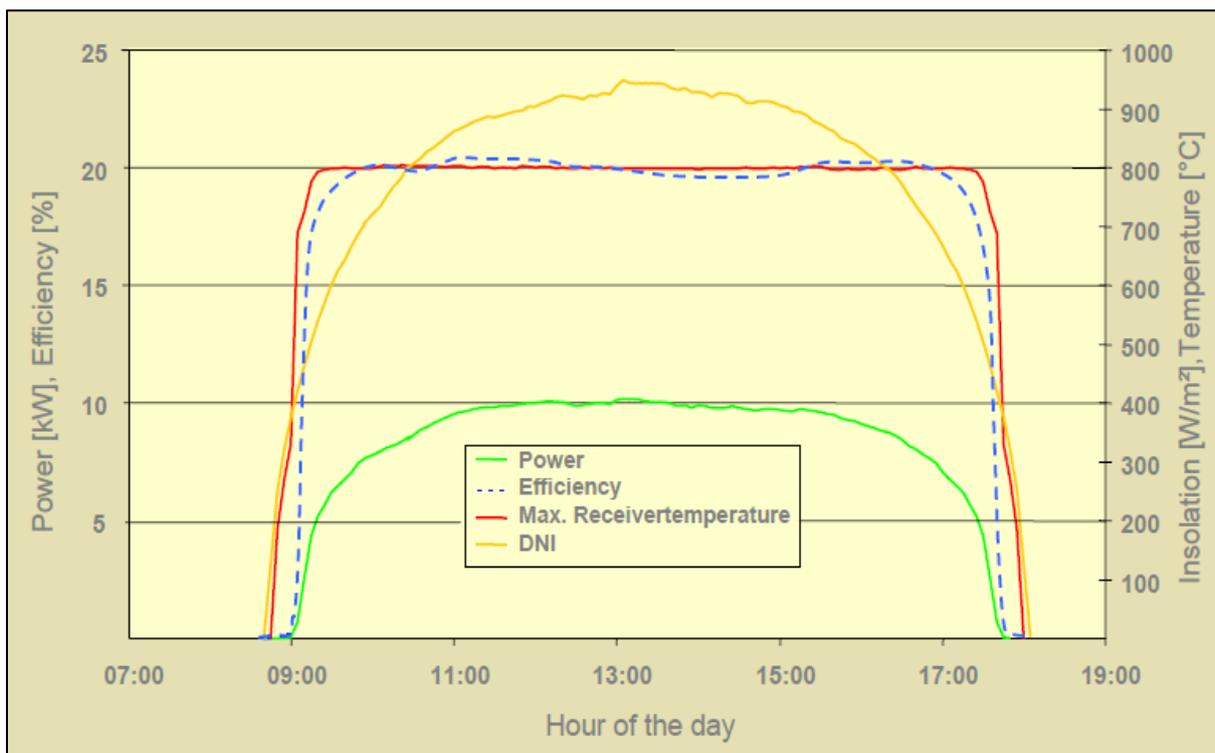


Figure 30 - Typical operation day in Spring (Measured on Feb. 2005 in Seville) [18]



Figure 31 – Eurodish – WF pressure in the engine [20]

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